

The WAKE4D Platform with the Deterministic wake Vortex Model (DVM) and the Probabilistic wake Vortex Model (PVM): Description and Applications

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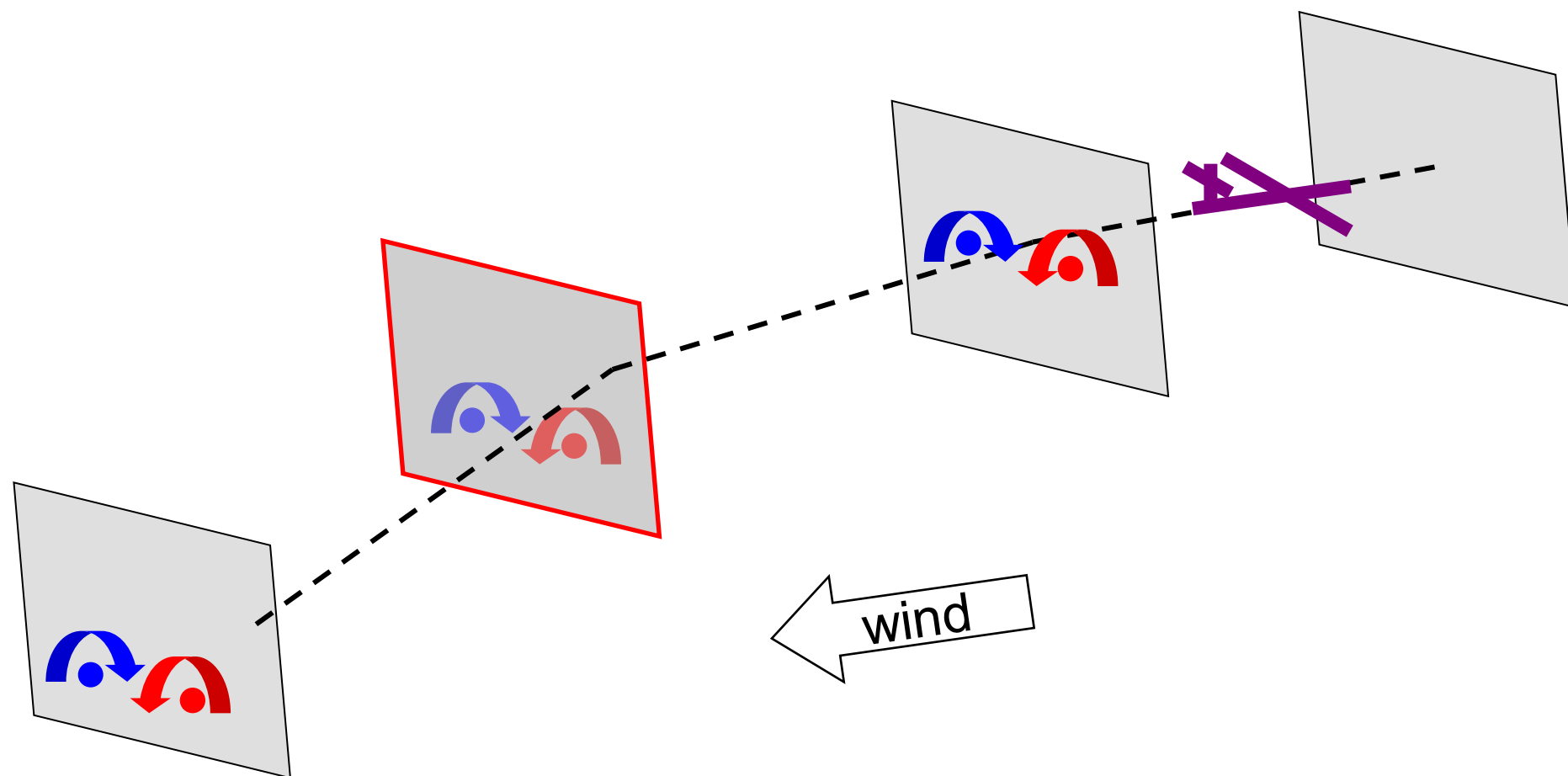
WakeNet3-Europe Specific Workshop
“Operational Wake Vortex Models”, UCL, 7-8 November 2011

Outline

- Presentation of the WAKE4D platform and its subcomponents DVM/PVM
 - Description of the DVM
 - Description of the PVM
 - Description of the WAKE4D
 - Description of the “trajectory generator tool” (Pre-Processor)
- Examples of application of the WAKE4D-DVM + Pre-Processor
 - 3-D wake of a typical approach to the Marseille-Provence airport with two wind profiles
 - Induced velocity field evaluation for a typical take-off
 - Induced velocity field evaluation taking into account the Crow instability
- Examples of application of the WAKE4D-PVM
 - Investigation of a Time-Based Separation (TBS) concept
 - Investigation of potential wake vortex encounter (WVE) for parallel runways

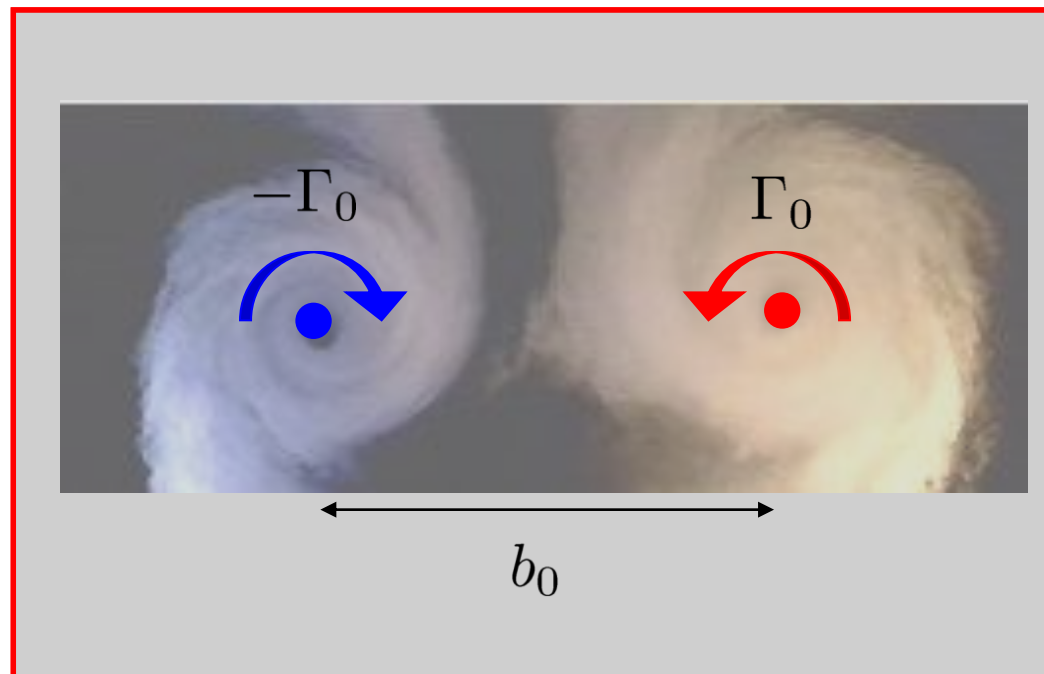
WAKE4D: “3-D space + time” wake vortex prediction platform

- It uses as **input**:
 - the **a/c trajectory**
 - the **met** conditions.
- The computational domain is divided in various “**computational gates**” crossing the flight path.
- The aircraft crossing one of the gates generates a **pair of wake vortices** (WV) that are **transported** (also by the headwind) and also **decay**.



The Deterministic wake Vortex Model (DVM)

- The DVM forecasts, in **real-time**, the WV behavior (transport and decay) in one **computational gate**, using simplified **physical models**.
- The initial wake is computed using the a/c characteristics (position, mass, TAS, wingspan, lift distribution, and flight angles) when the aircraft crosses the gate.

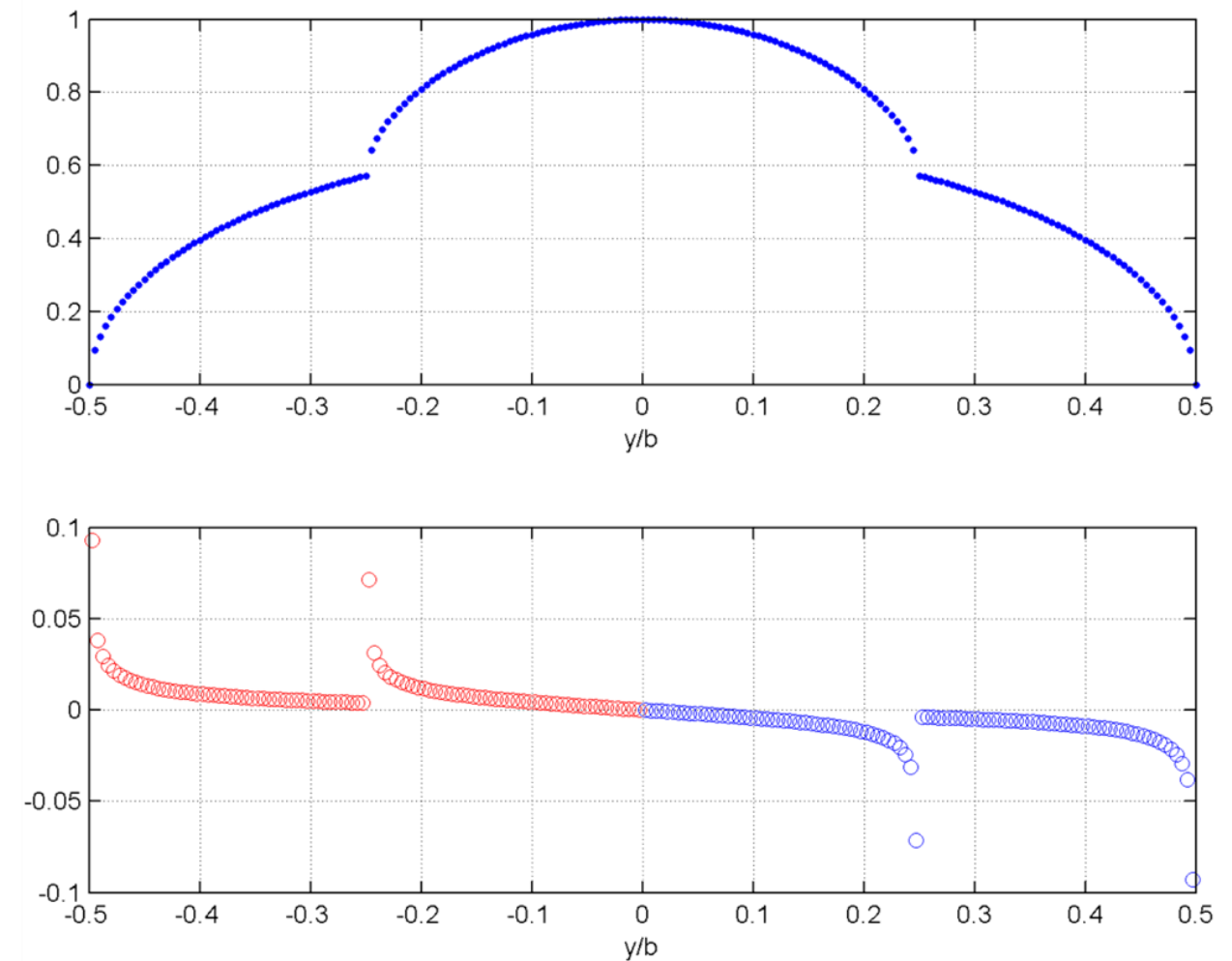
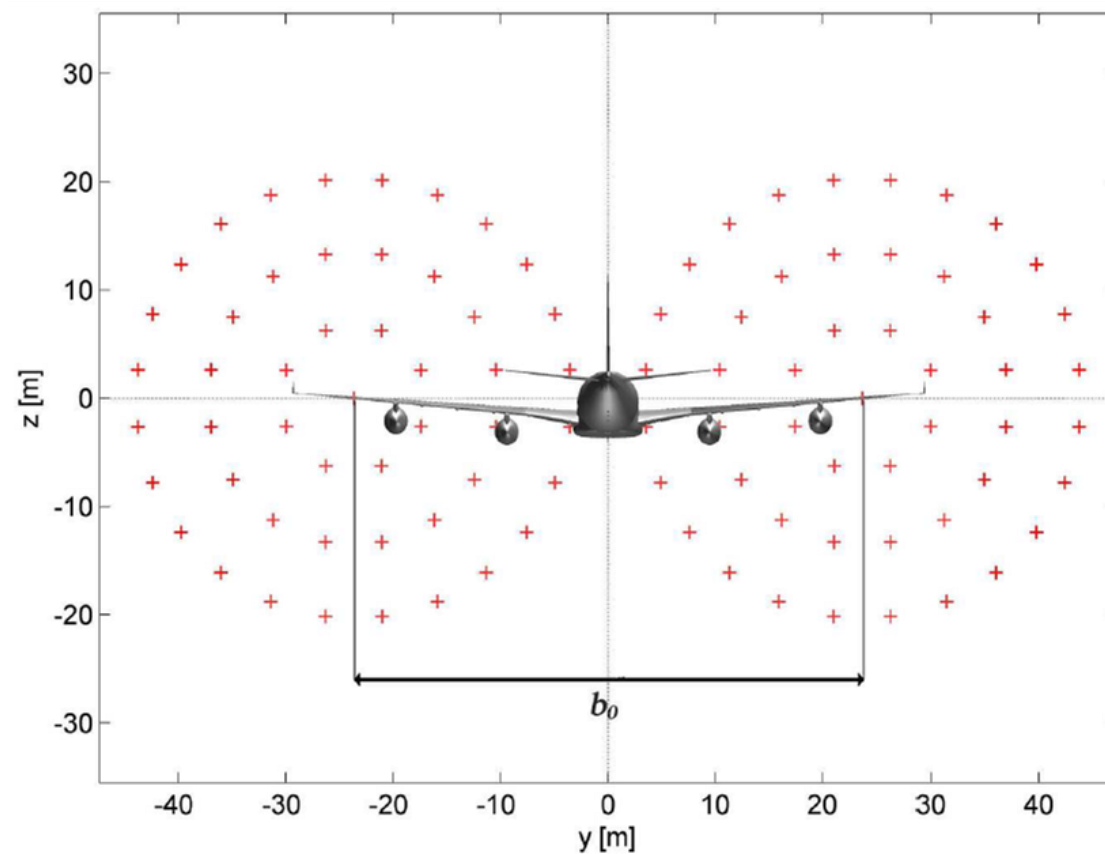


$$\begin{aligned}
 b_0 &= s b \\
 \Gamma_0 &= \frac{Mg}{\rho U_\infty b_0} \\
 V_0 &= \frac{\Gamma_0}{2\pi b_0} \\
 t_0 &= \frac{b_0}{V_0}
 \end{aligned}$$

- Each wake vortex is represented by a **vortex particle** with a chosen circulation distribution profile:
 - the Burnham-Hallock model (= low-order algebraic model),
 - the high-order algebraic model, or
 - the two-scales Proctor-Winckelmans model

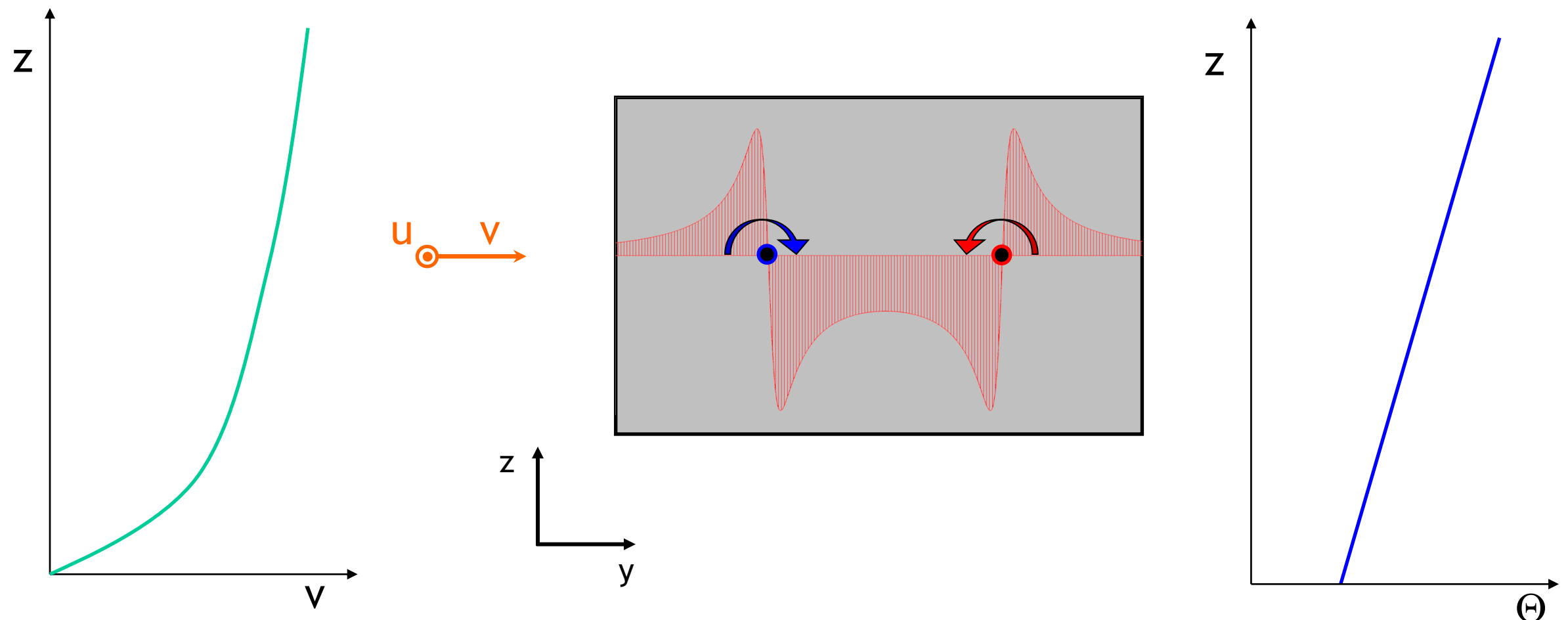
Other available wake initialization options

- Near wake model
 - 2 rolled-up vortices with the Proctor-Winckelmans distribution discretized using n layers of particles (example $n=3$)
- Vortex sheet with various circulation distributions
 - Example: double elliptical distribution



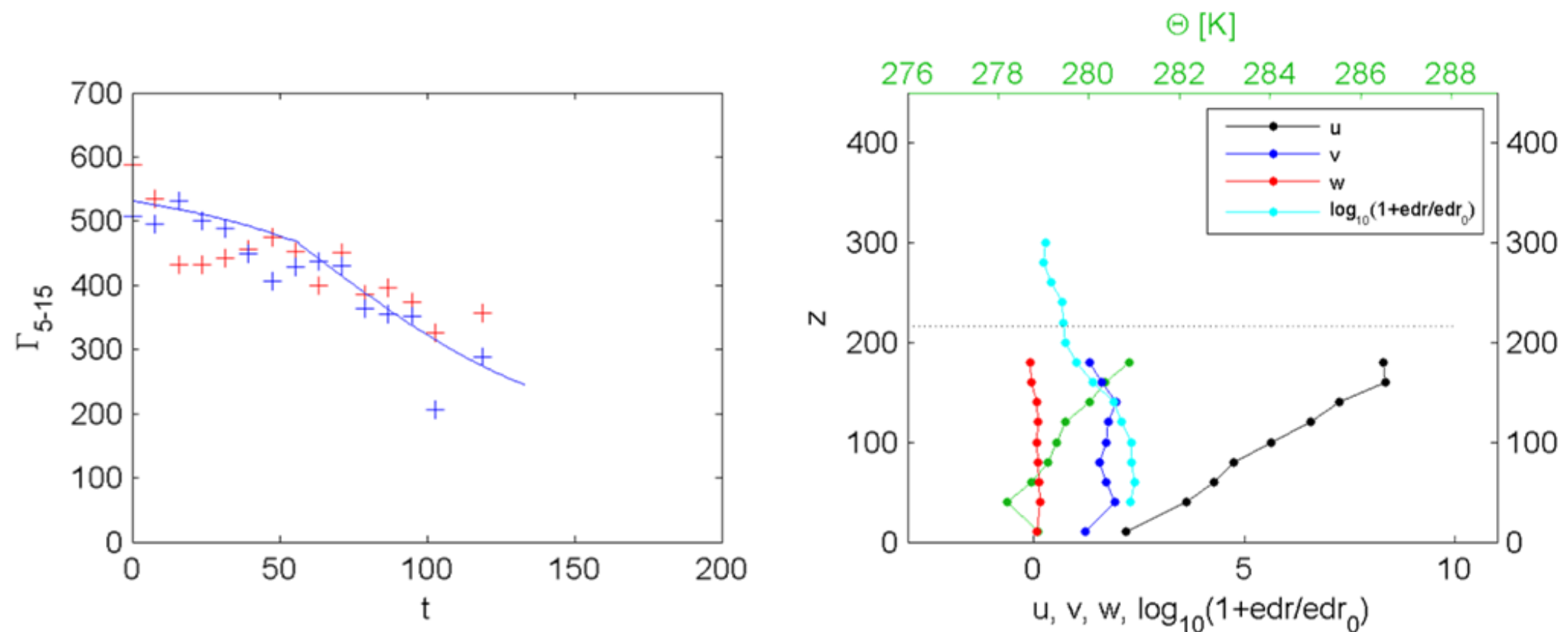
The physical transport models used in the DVM

- **Self-induced velocity**: Biot-Savart law (also using image particles for NGE modeling)
- **Wind convection**: by both the axial and lateral components evaluated at the altitude of WV evolution
- **Wind shear**: the shear of the wind profile is computed and the tilting effect is modeled
- **Stratification effects**: the stratification level, N^* , is computed from the potential temperature profile and the thermally induced rebound is modeled



The physical models used in the DVM

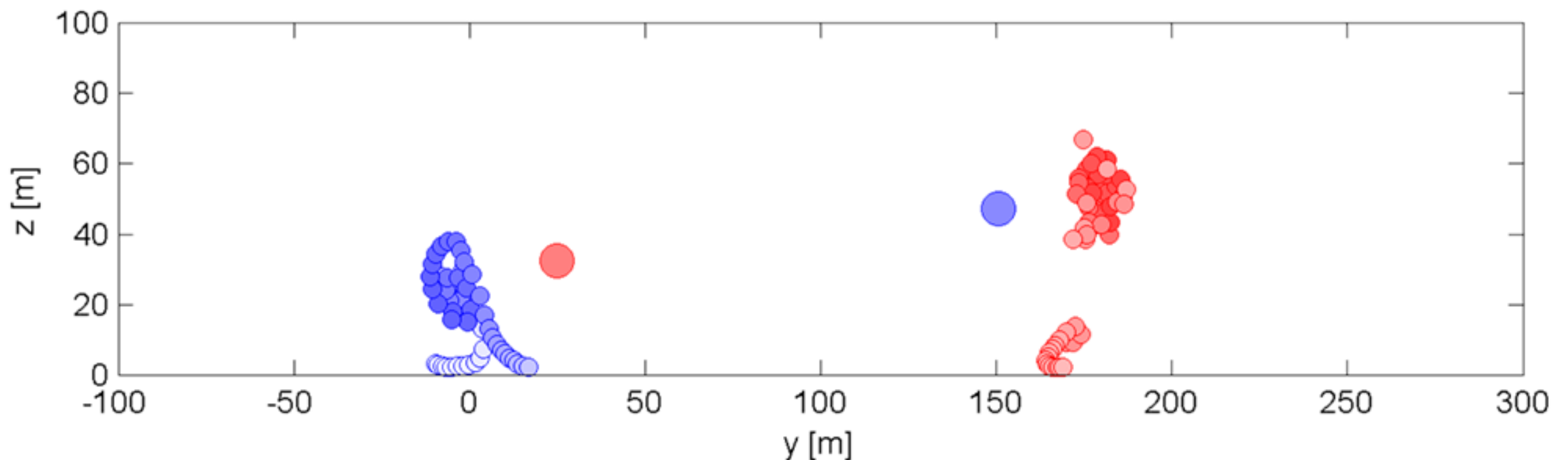
- Circulation decay
 - **Two phase decay**: slow decay phase followed by a rapid decay phase
 - Accumulated **“time-to-demise”** approach: starting time of the fast decay phase $t_d(\text{EDR}, N^*)$
 - The decay rate of both phases also **depends on EDR and N^***
 - Possibility to also use a TKE-based decay model or the APA decay model using EDR



- Crow instability model
 - Model of the space developing Crow instability amplitude for predicting the WV deformation
 - Based on the time-to-demise and thus the **atmosphere turbulence level**

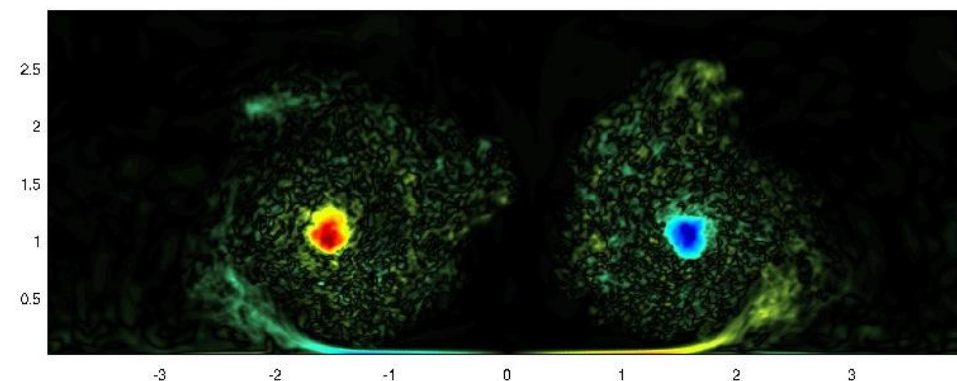
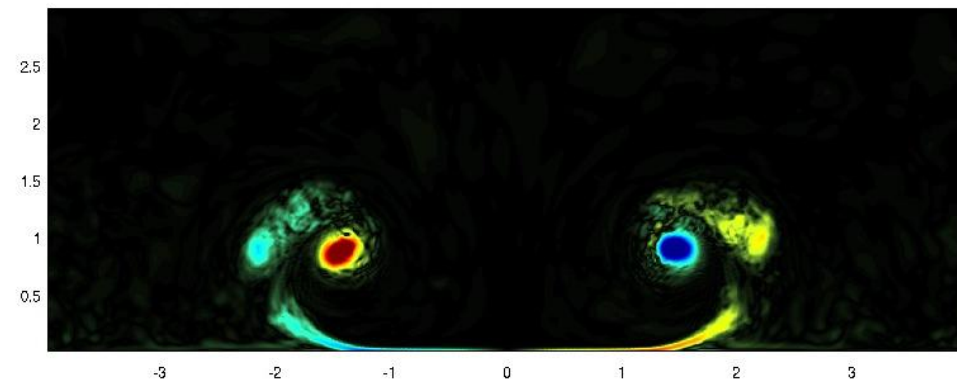
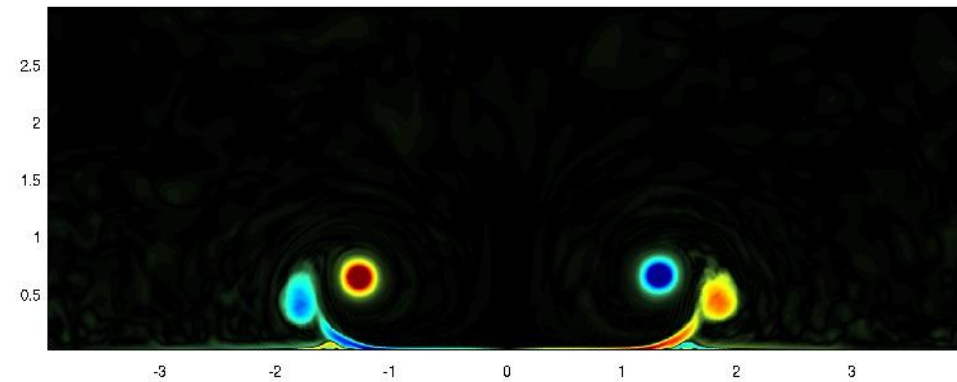
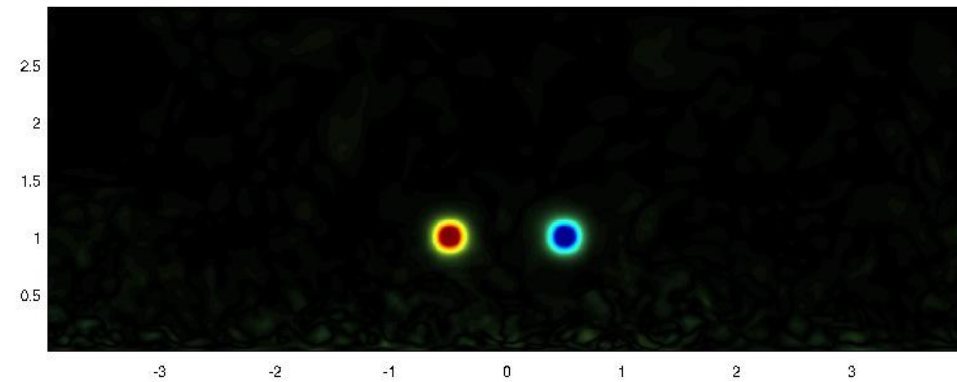
DVM In Ground Effect (IGE) modeling

- Generate **new « secondary » vortex particles** close to the ground, to model the ground-generated boundary layer due to no-slip
- Those particles also **dynamically separate** from the ground
- They interact with the primary vortices and induce the **rebound**
- There is a **redistribution** of the secondary particles when the primary vortex has bounced
- An additional **IGE decay model**, based on a “Particle strength exchange”, enhances the decay of the vortices IGE after rebound.

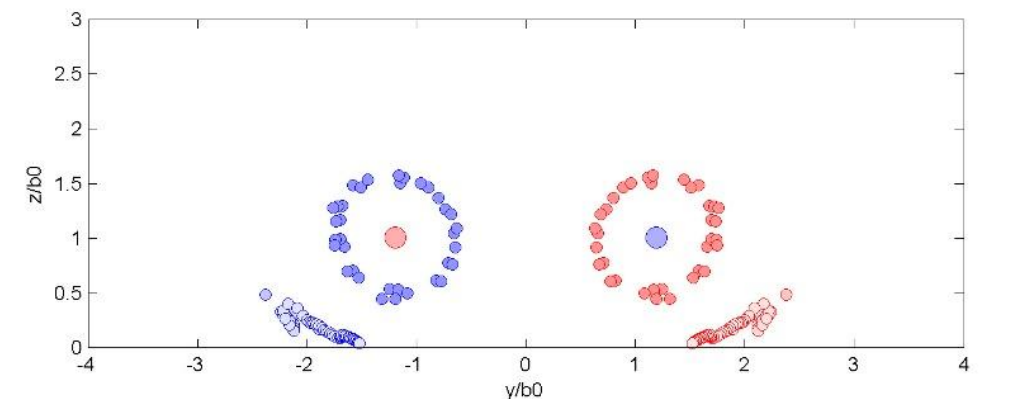
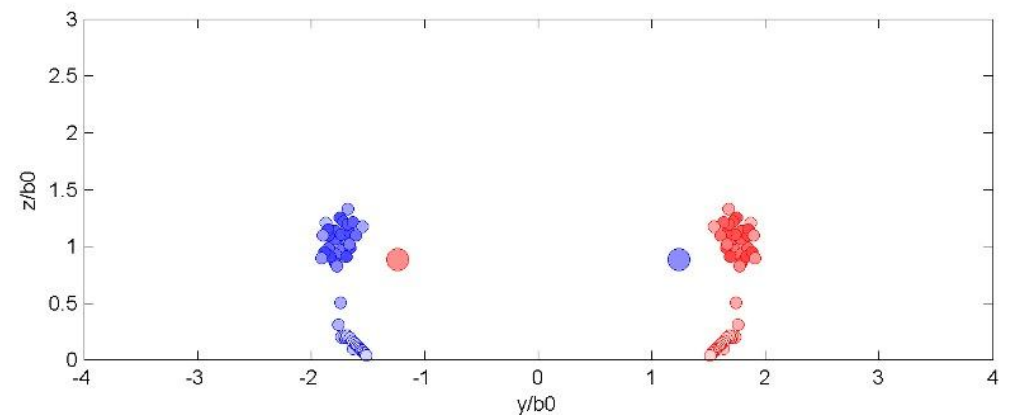
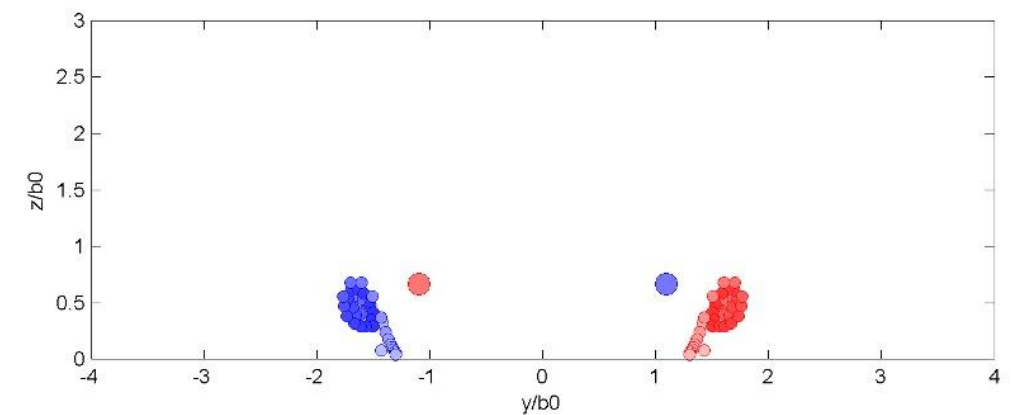
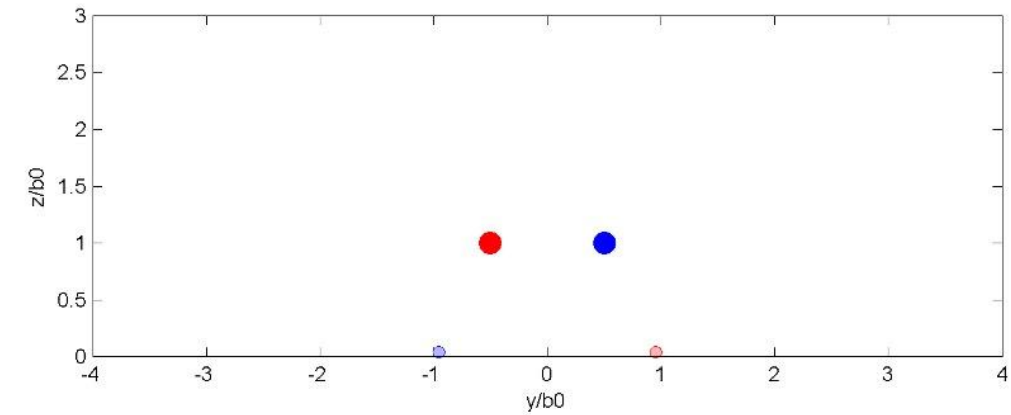


Comparison between LES and DVM results: case IGE with headwind

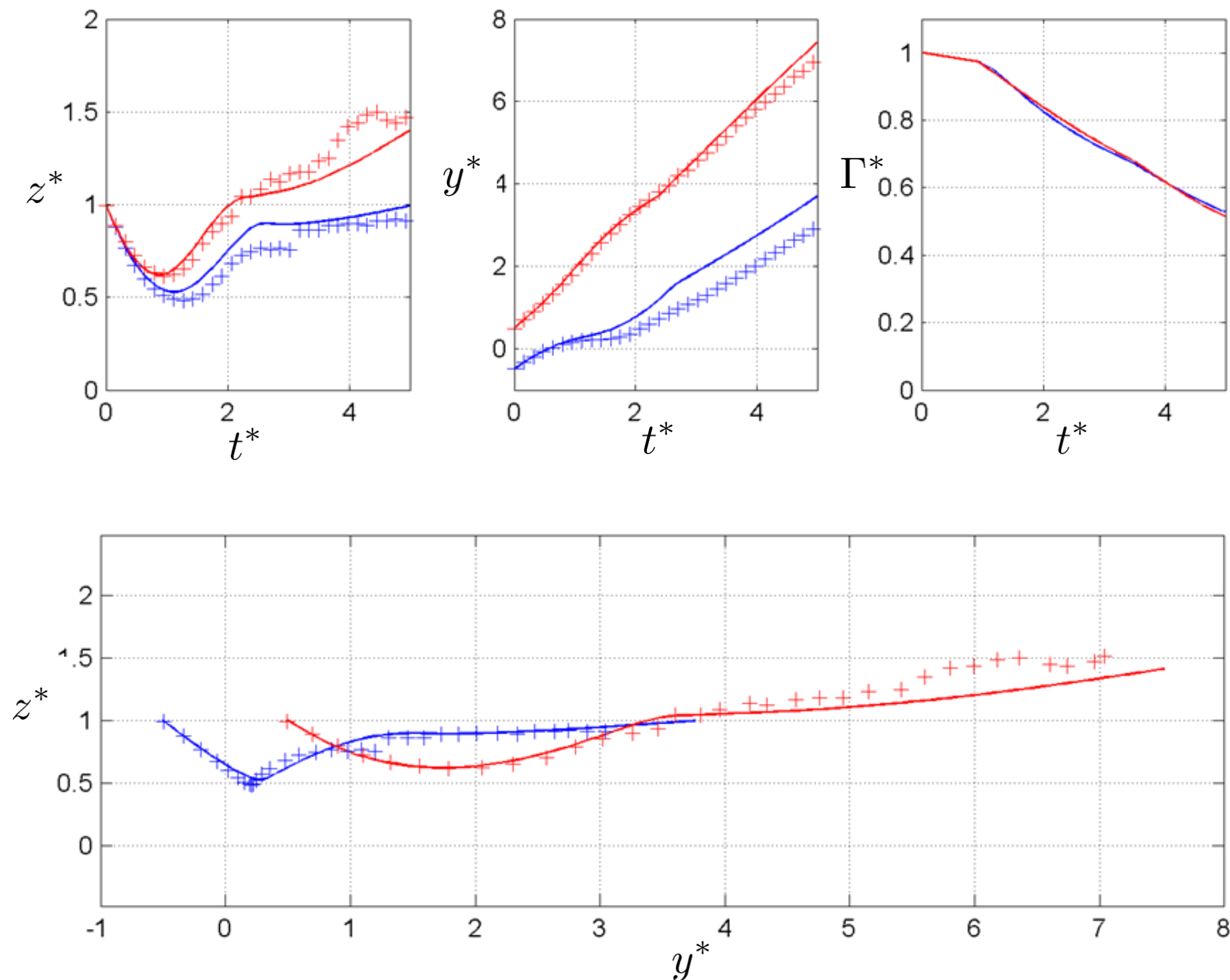
Mean axial vorticity field as computed by a LES



Particles used in the DVM to model WV IGE



Example of comparison between LES and DVM results: case IGE with crosswind



$$z^* = z/b_0$$

$$y^* = y/b_0$$

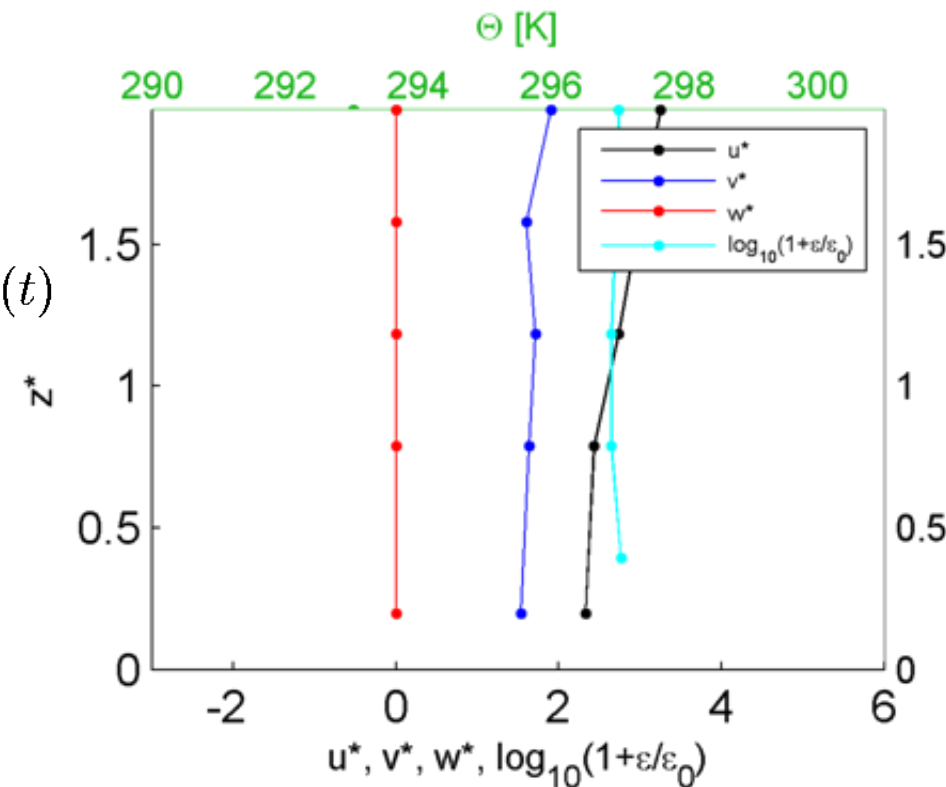
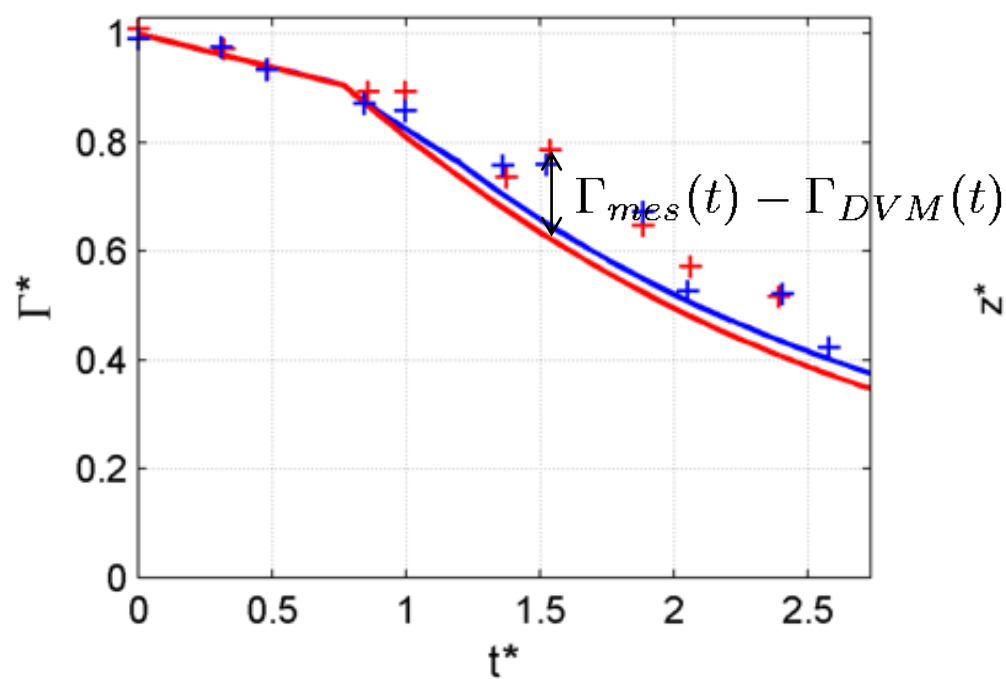
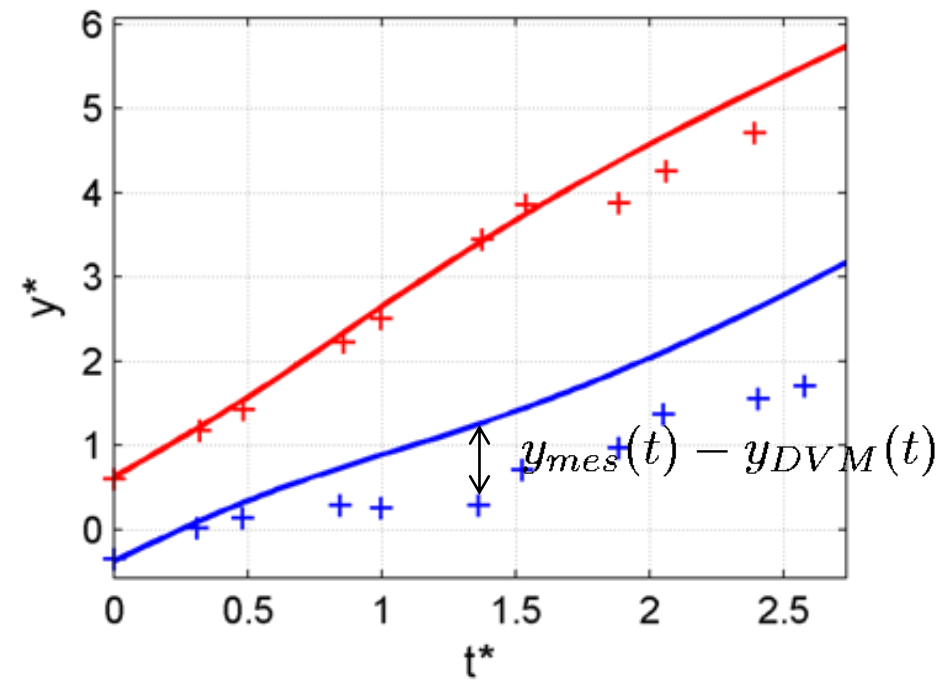
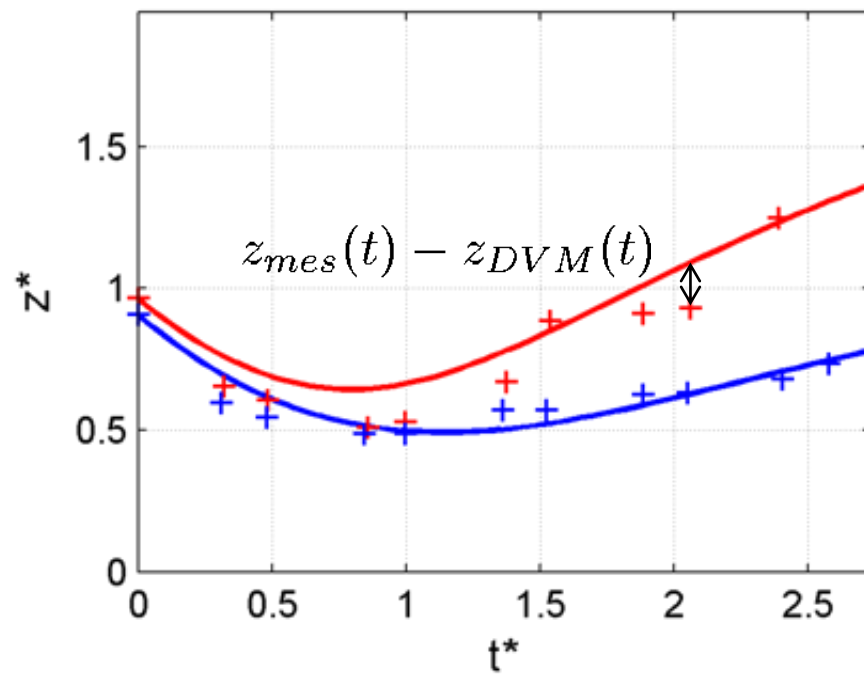
$$\Gamma^* = \Gamma/\Gamma_0$$

$$t^* = t/t_0$$

Crosses: LES
 Solid curves: DVM
 Blue: port vortex
 Red: starboard vortex

Example of comparison between DVM results and measurements

LIDAR measurement campaign of WV generated by heavy aircraft in ground proximity (WakeFRA 2004)



$$z^* = z/b_0$$

$$y^* = y/b_0$$

$$\Gamma^* = \Gamma/\Gamma_0$$

$$t^* = t/t_0$$

Crosses: measurements

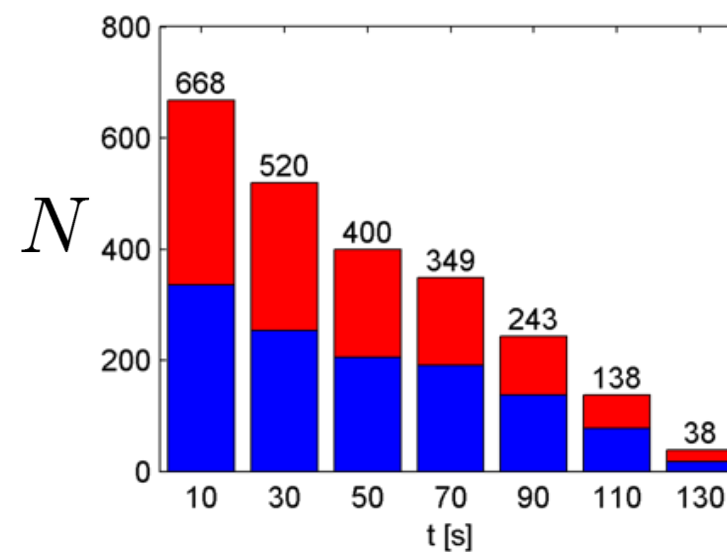
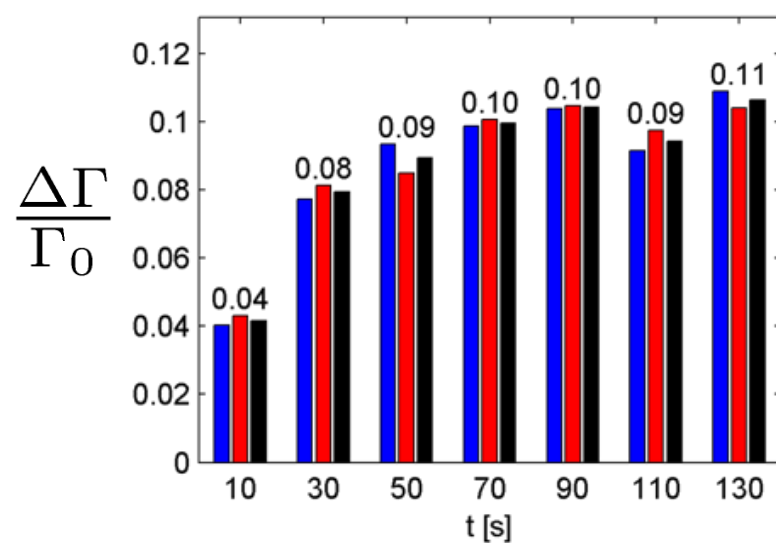
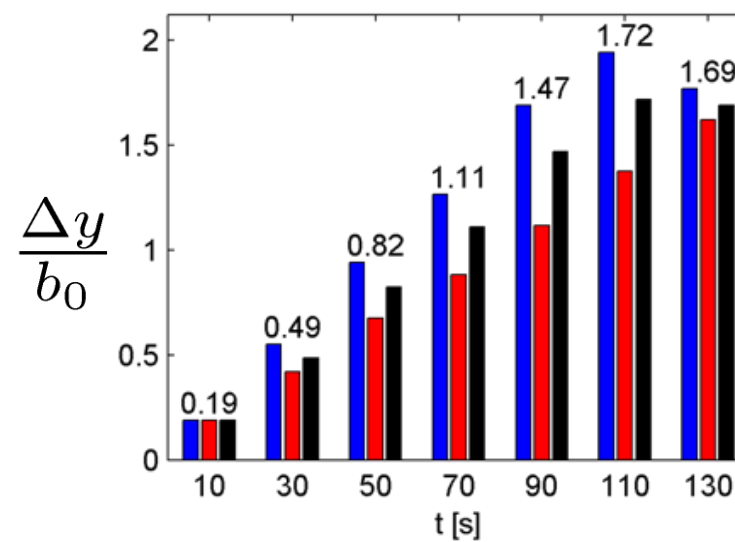
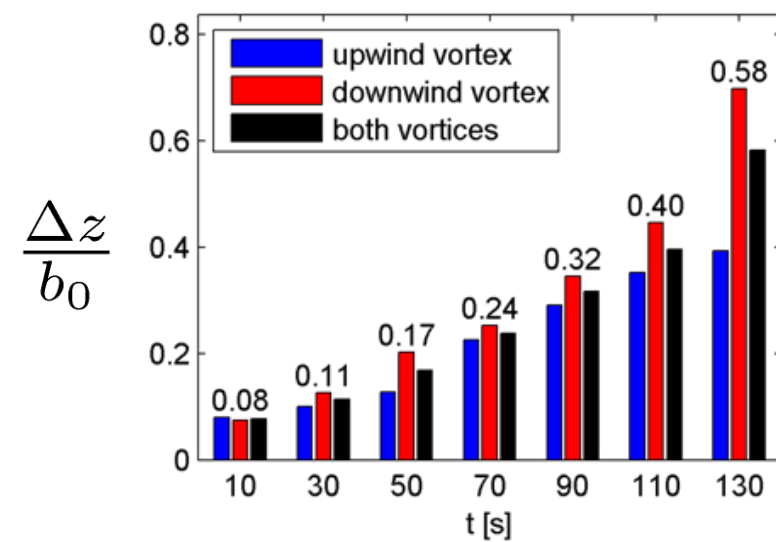
Solid curves: DVM

Blue: port vortex

Red: starboard vortex

Example of assessment of the DVM against measurements (using WakeFRA 2004)

Histograms of the deviations between measurements and predictions as a function of time (t) for the vertical (Δz) and lateral (Δy) positions and for the circulation ($\Delta \Gamma$). The number of cases in each time frame (N) is also provided.



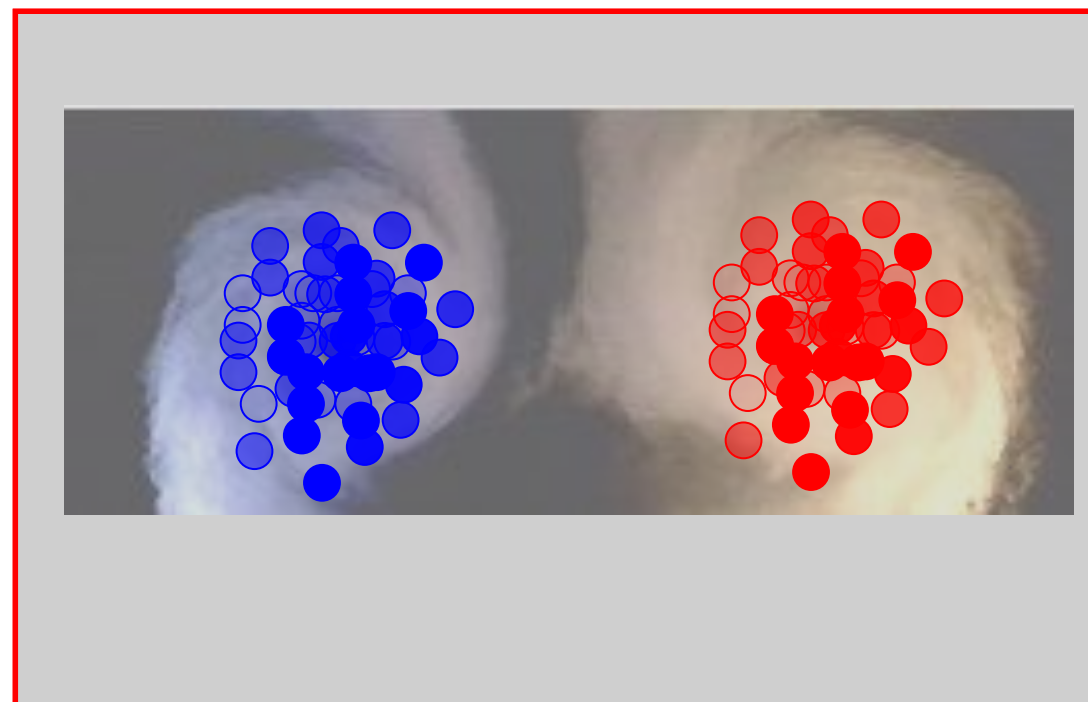
$$\Delta z(t) = \sqrt{\frac{1}{N(t)} \sum (z_{mes}(t) - z_{DVM}(t))^2}$$

$$\Delta y(t) = \sqrt{\frac{1}{N(t)} \sum (y_{mes}(t) - y_{DVM}(t))^2}$$

$$\Delta \Gamma(t) = \sqrt{\frac{1}{N(t)} \sum (\Gamma_{mes}(t) - \Gamma_{DVM}(t))^2}$$

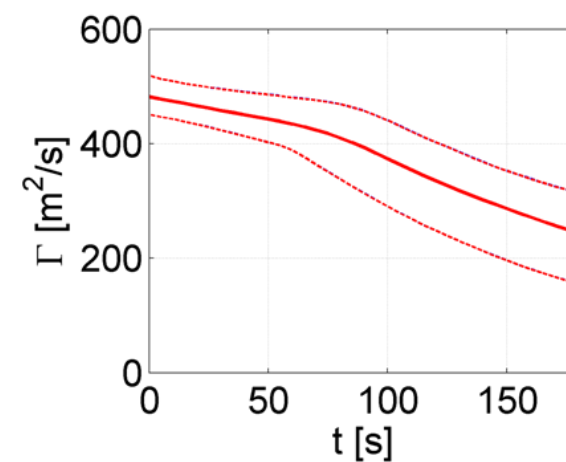
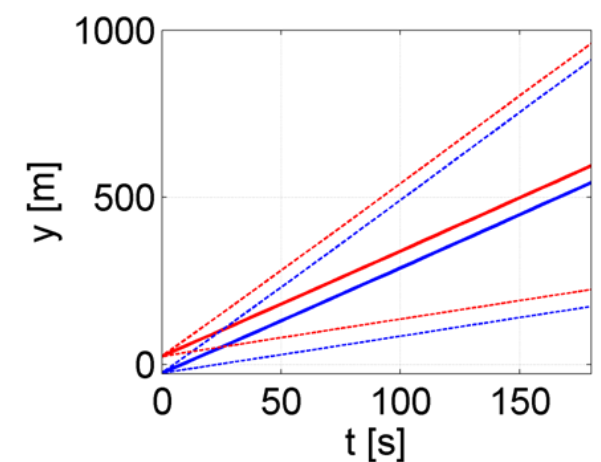
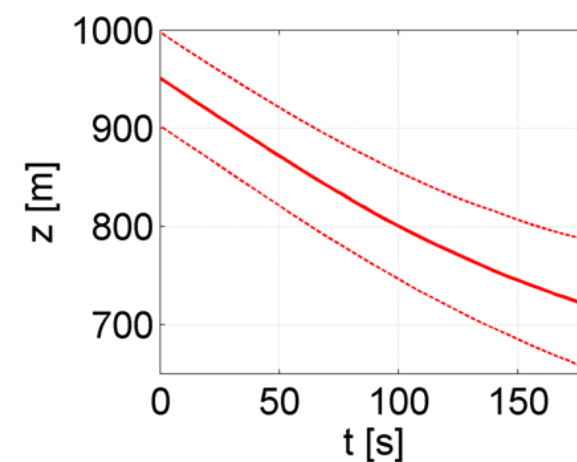
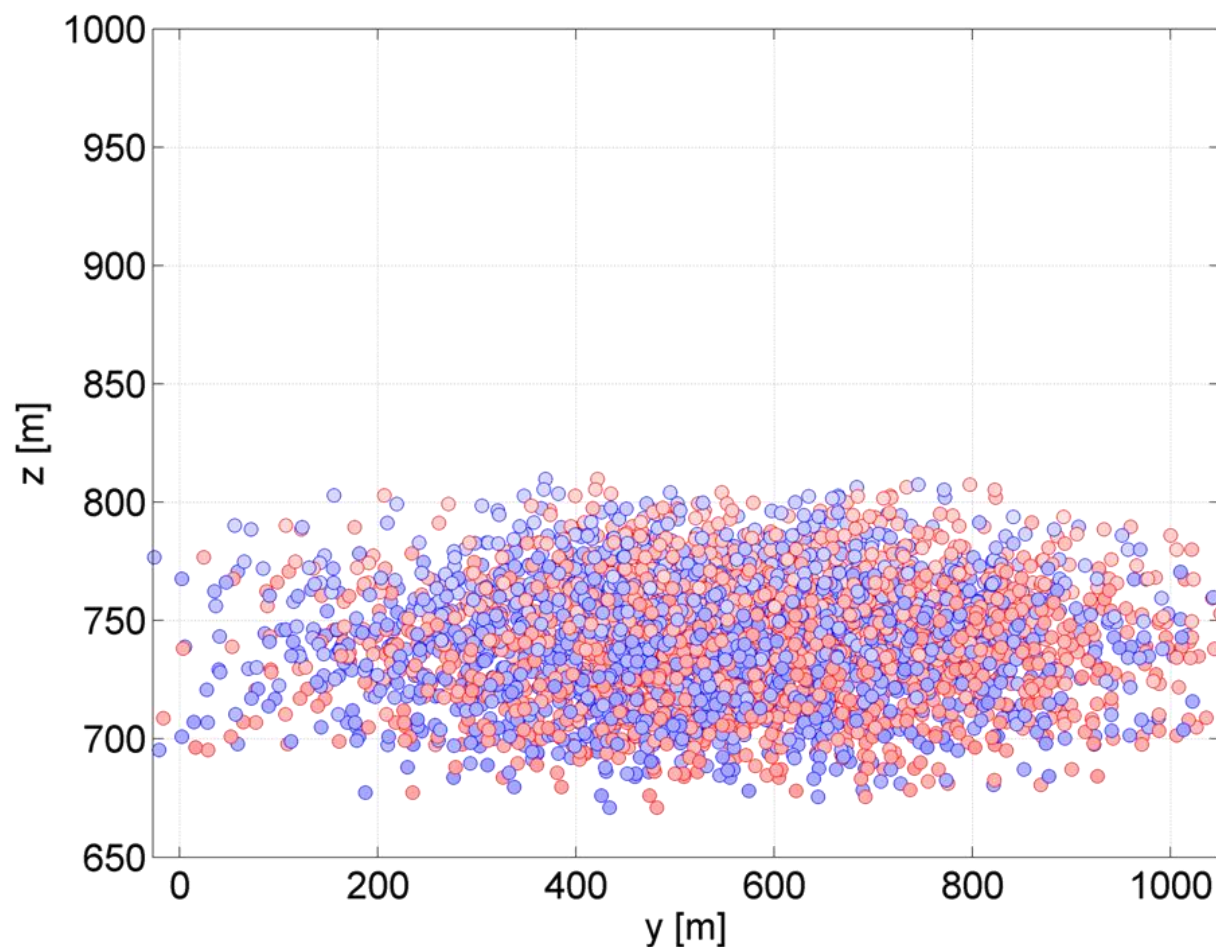
The Probabilistic wake Vortex Model (PVM)

- Probabilistic modeling and assessment of wake vortices is often what is **operationally required**.
- The PVM is an upper software layer, based on a **Monte-Carlo approach**, using the DVM as a sub-tool.
- For each PVM run, **many DVM runs** are performed, **with random variations** on the impact parameters (each one following its own distribution):
 - met conditions (natural variations and uncertainties)
 - a/c characteristics (uncertainties)
 - some coefficients of the physical models (calibration uncertainties)



PVM outputs

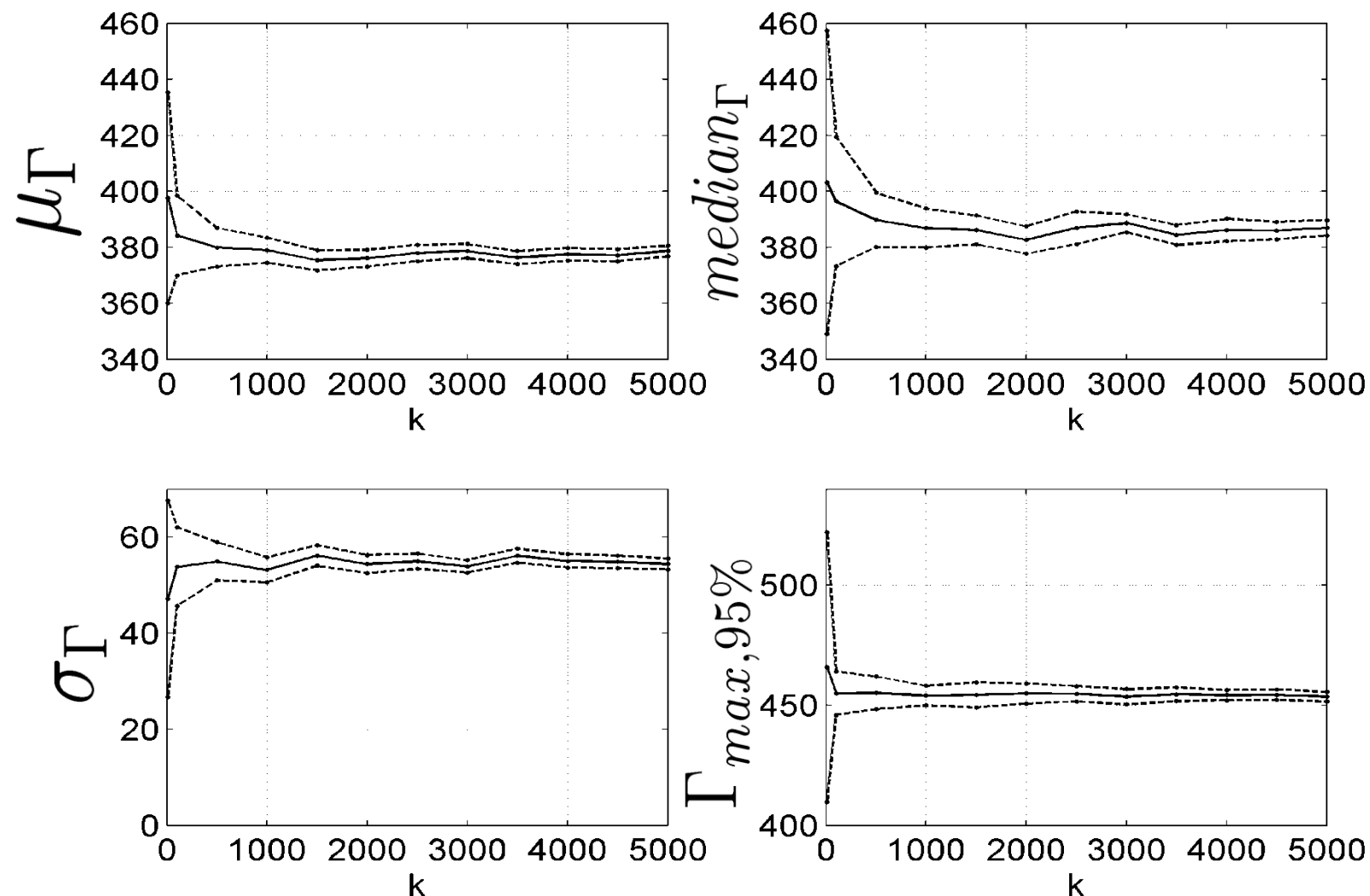
- One then obtains, in a computational gate at each time:
 - a set of vortex positions,
 - their associated circulations.
- The output distribution is not just a “simple image” of the input distribution
- A **statistical analysis** on the result sample can be performed to obtain:
 - PDF, mean, median, variance, percentiles, ...



Solid : median
Dash : 95 % envelopes

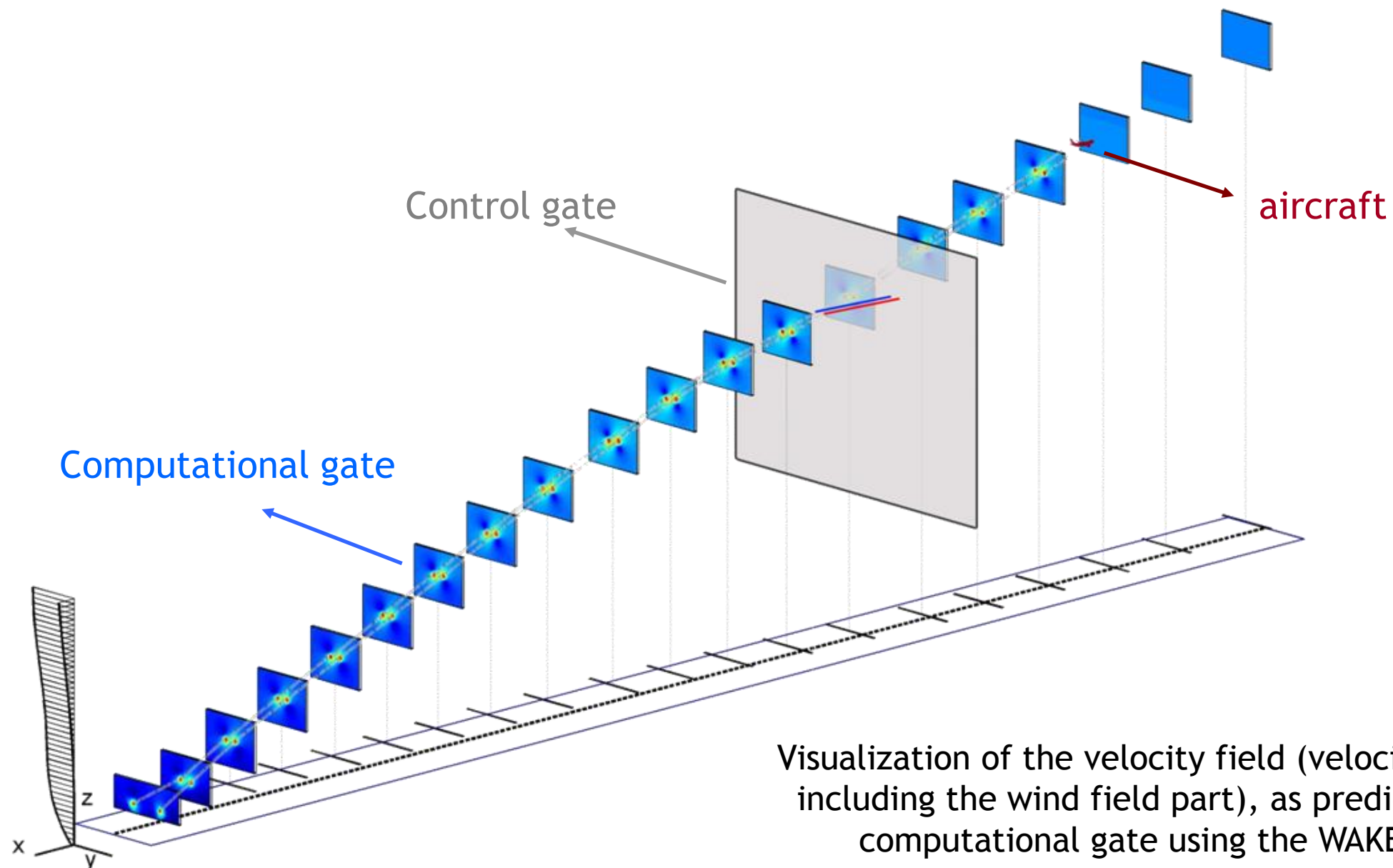
Bootstrap resampling technique

- Aim : obtain **conservative statistical** results while **limiting the number of DVM runs**
- The resampling technique provides an estimate of the variance of the statistics of the Monte-Carlo results: the “statistics on the statistics”
- Obtain PVM confidence envelopes that are sufficiently accurate and usable, using a moderate number of DVM runs.
- Obtain a **PVM approach** that is **computationally efficient**; hence also usable for real-time systems.



WAKE4D-DVM and WAKE4D-PVM

- The platform can be run deterministically (using the DVM in each gate) or probabilistically (using the PVM in each gate).
- From the 3-D “gate by gate” DVM (resp. PVM) computations, one can rebuild the 3-D wake (resp. envelope of the wake).
- “Control gate”: interpolation in a fixed plane (similar to a LIDAR scanning plane).
- “Control box”: recording of the vortices present in a box as a function of time.

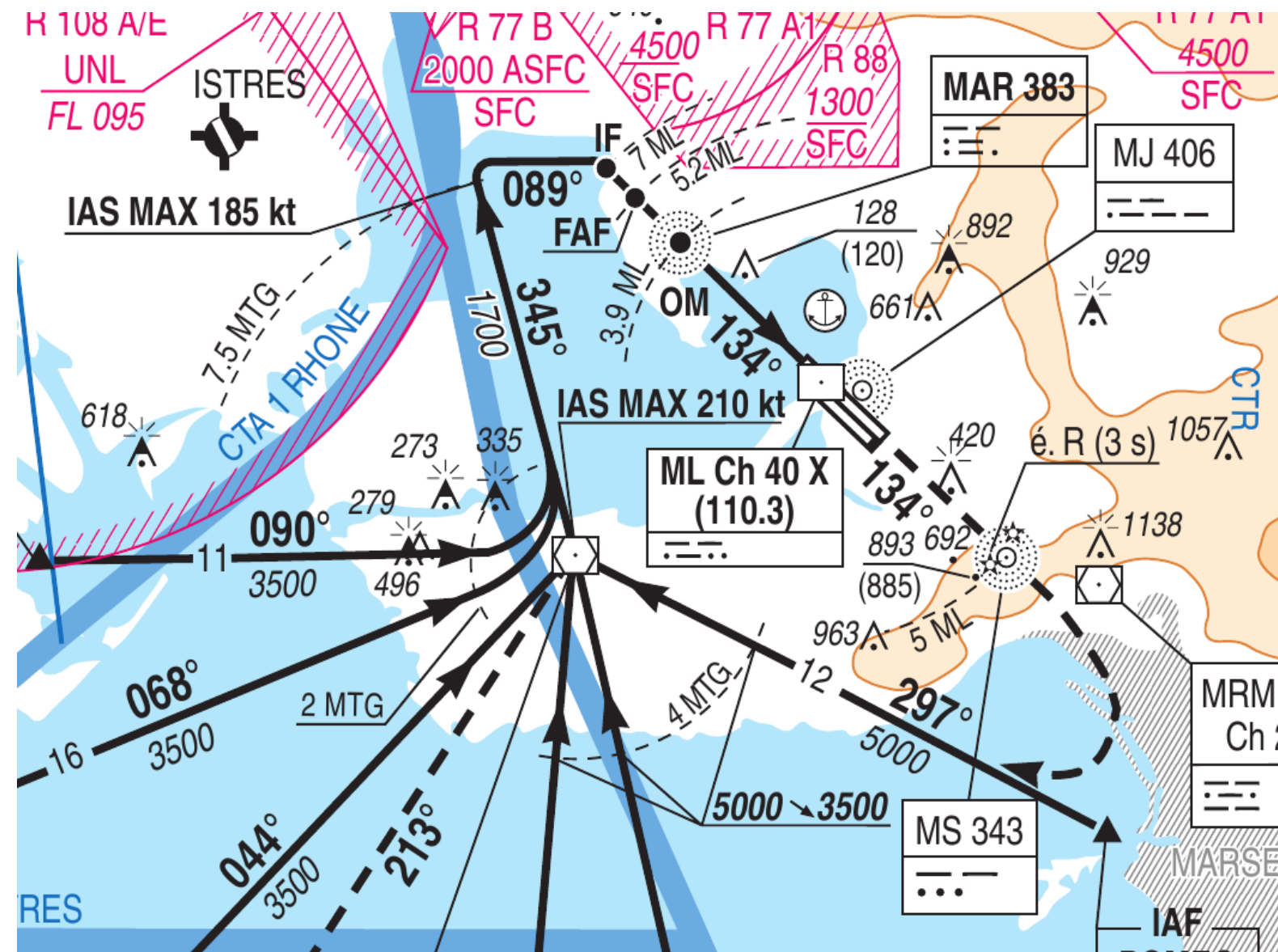


The Pre-Processor tool

- Helps the user in the generation of the a/c trajectory
- Takes several **waypoints** as inputs and the trajectory type (**straight line**, **left/right turn**) between each of them
- The pre-processor computes all inputs needed by the WAKE4ED as a function of wind, TAS and aircraft mass:
 - Position evolution: $x(t)$, $y(t)$, $z(t)$
 - Bank angle evolution: $\phi(t)$
 - Relative azimuthal and elevation angles evolutions: $\psi_r(t)$, $\gamma_r(t)$
- Left/right turn: constant rate of turn ω . Typically, we use $\omega = 3[deg/s]$

The Pre-Processor tool: application for the approach to the Marseille-Provence airport

- Aircraft : B747-400
 - MLW : 295,000 kg
 - $b=64.4$ m
 - $s= 0.75$ [-]
- Met conditions :
 - **Wind:** 15kts at $z=10$ m;
log profile in the TBL;
Ekman spiral
 - **Turbulence:** EDR is fixed by wind
within TBL and is equal to
 $1e-3$ [m^2/s^3] above
 - **Stratification:** $N = 0.01$ [$1/s$]
($N^* = 0.29$)



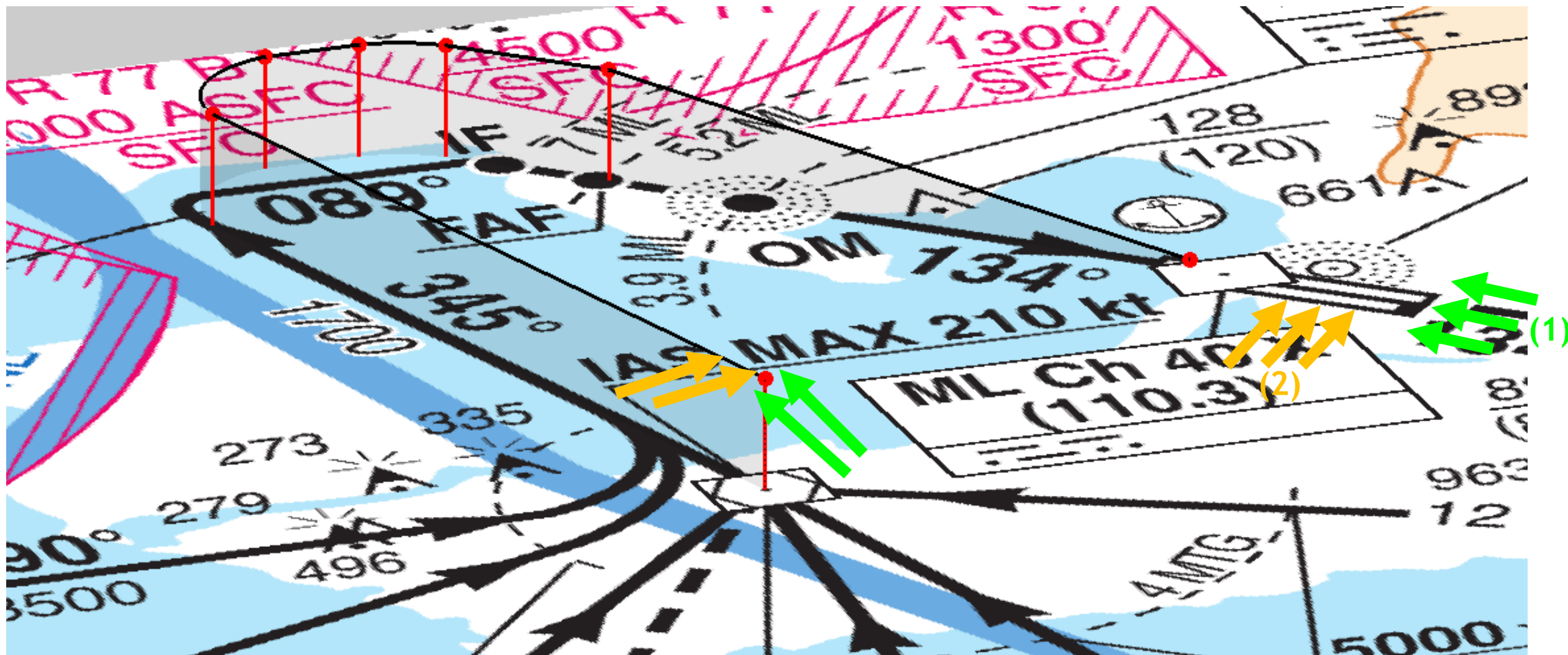
The Pre-Processor tool: application for the approach to the Marseille-Provence airport

- Two wind directions are investigated:

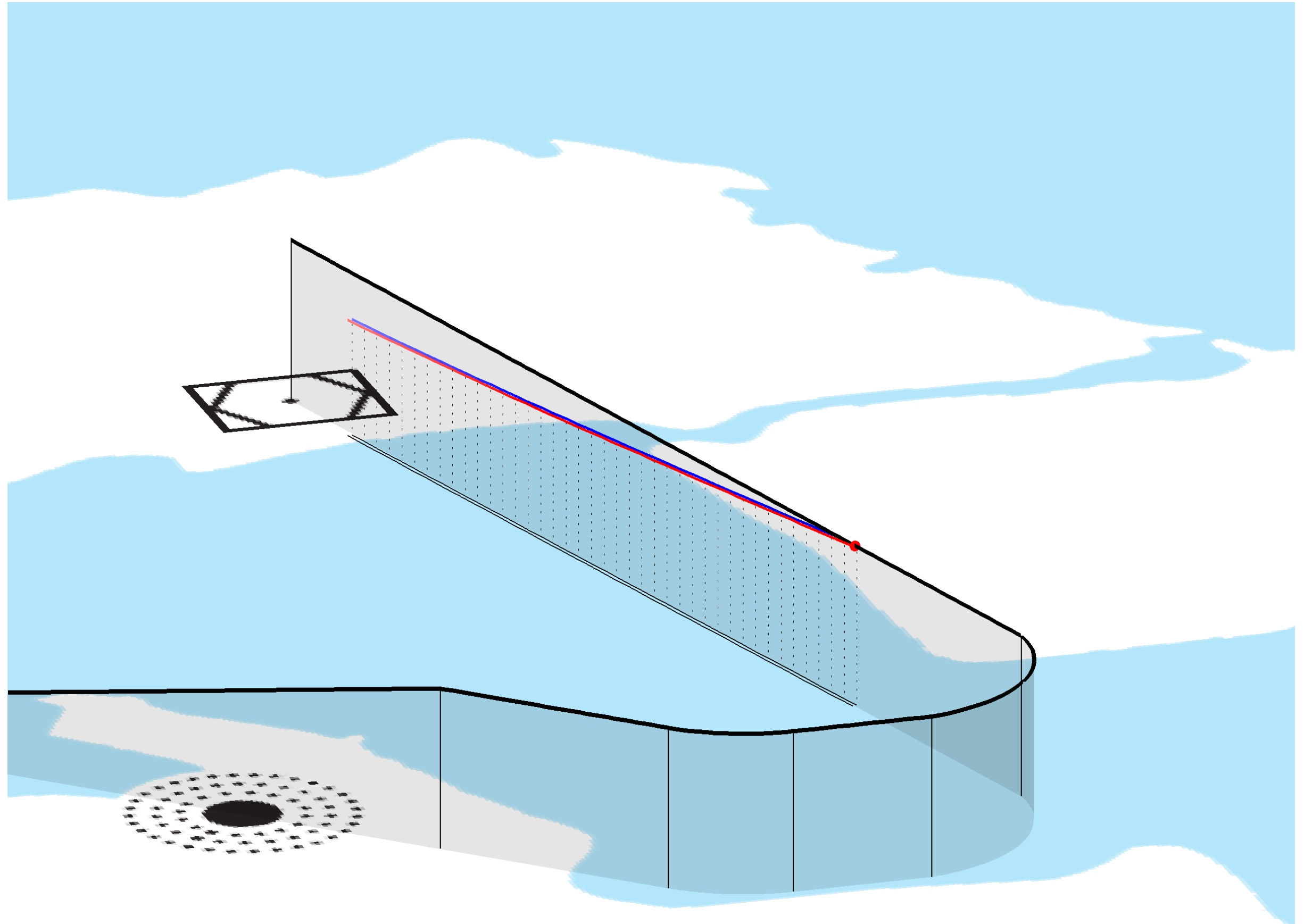
(1) Pure headwind on the runway (S-E wind)

(2) Pure crosswind on the runway (S-W wind)

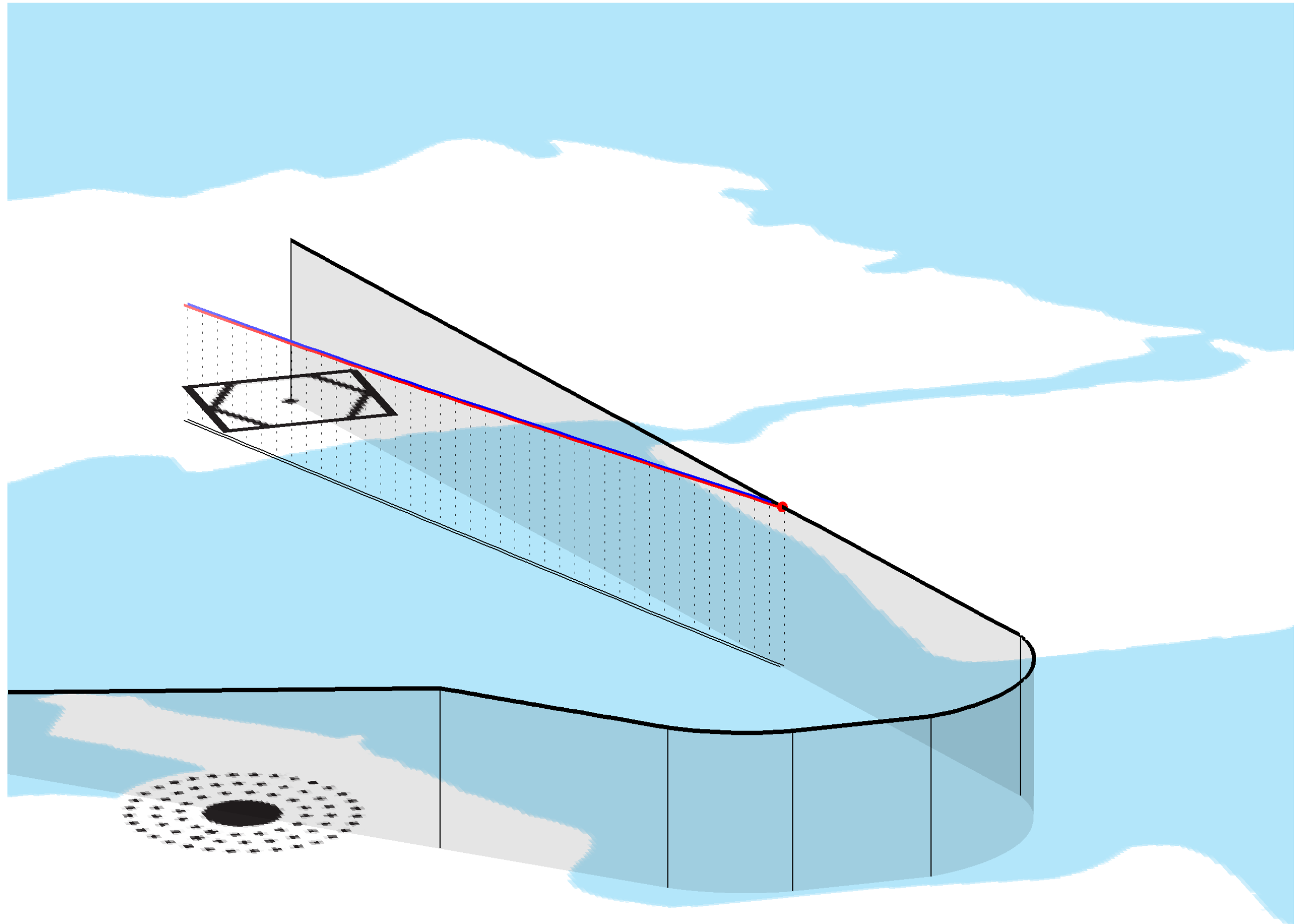
- 7 **waypoints** are needed (4 straight lines + 2 right turns)



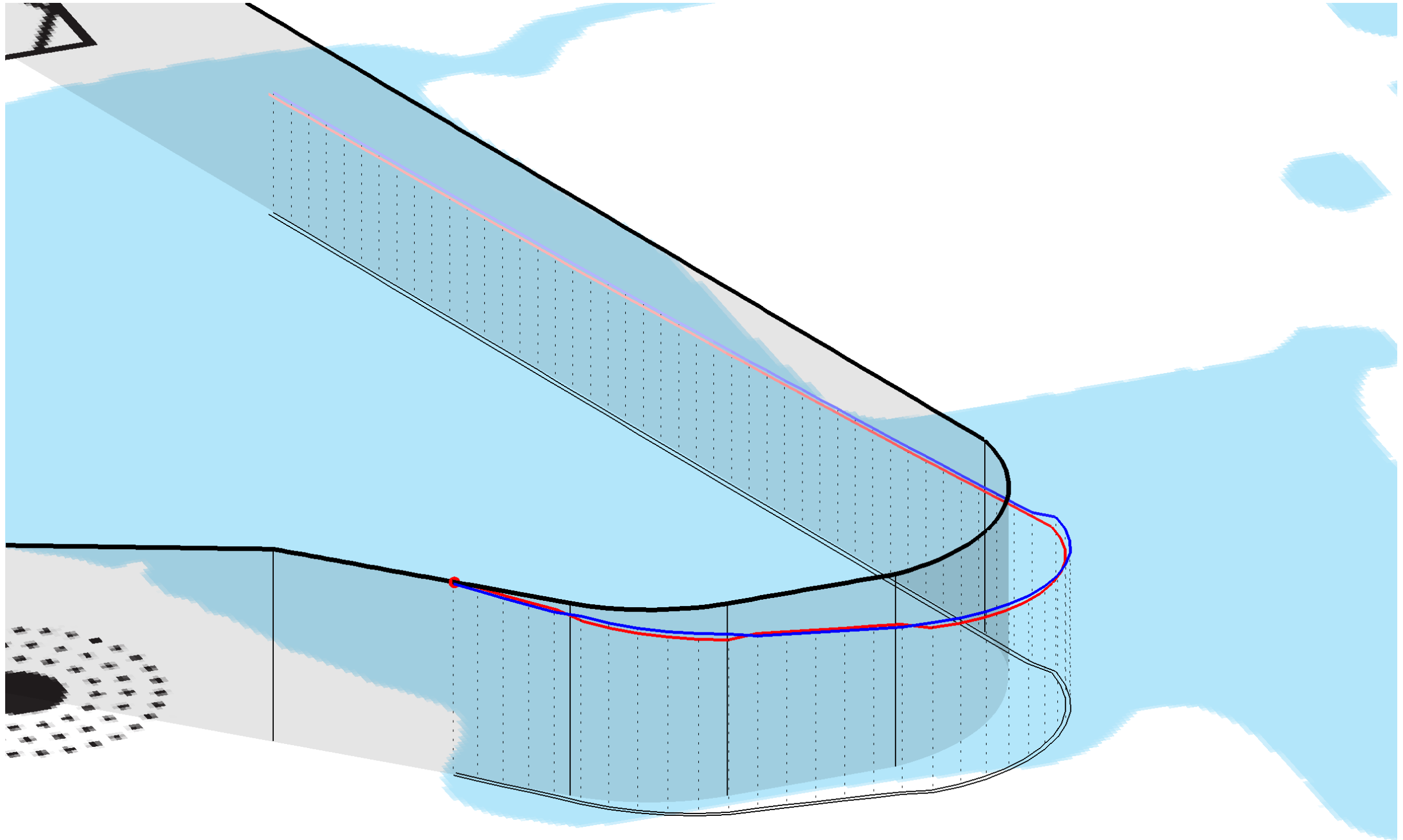
Wake at time $t=100$ s for S-E wind



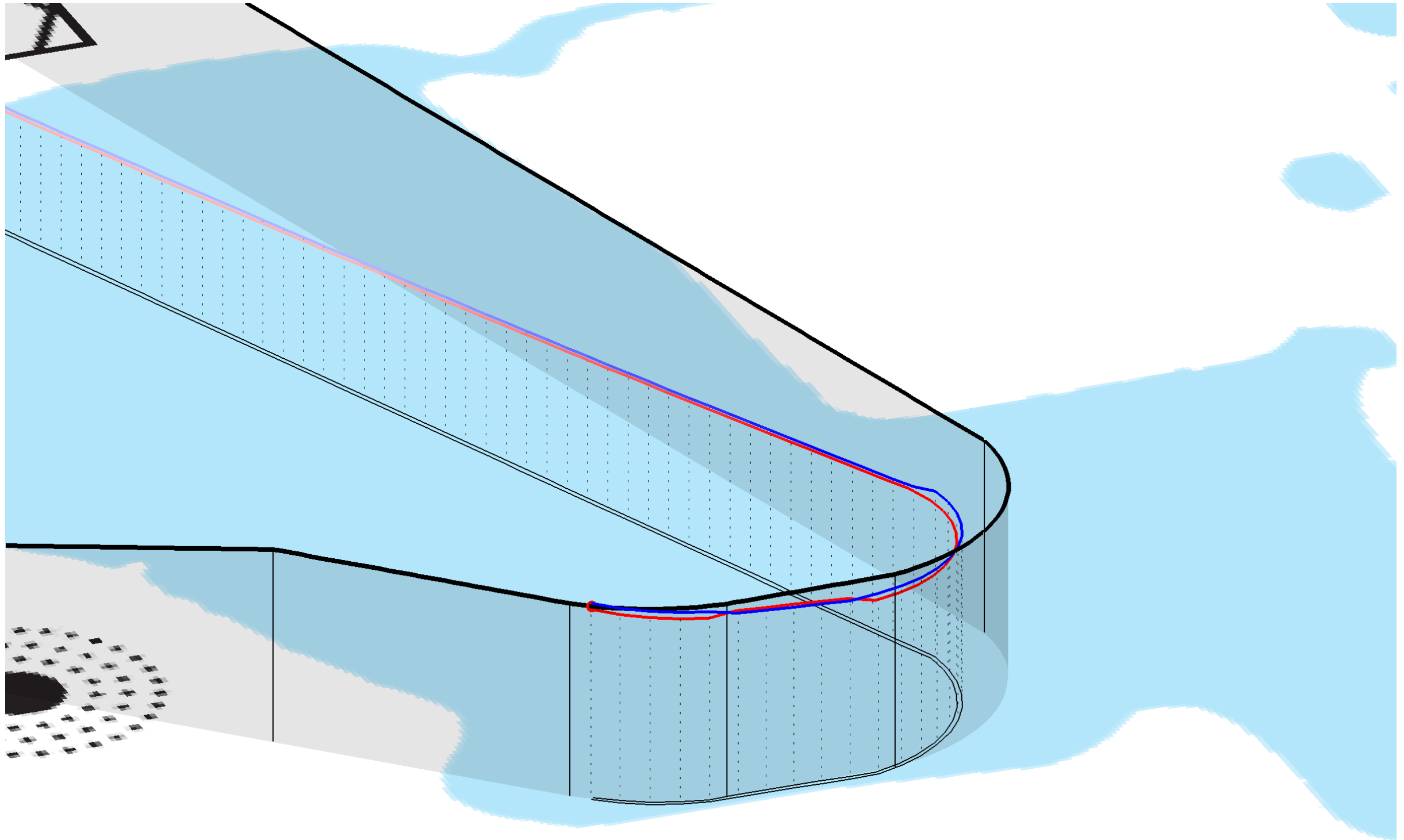
Wake at time $t=100$ s for S-W wind



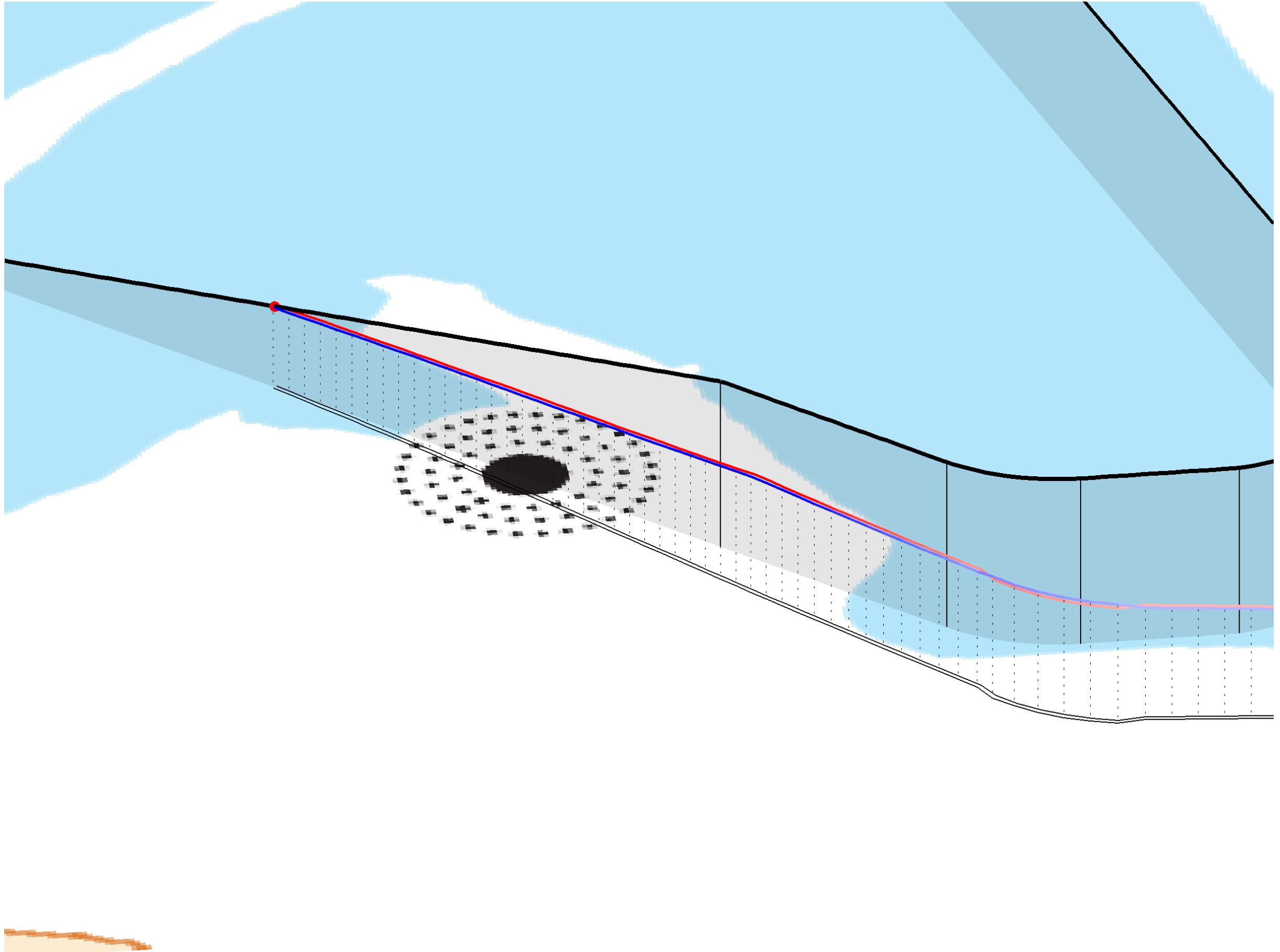
Wake at time $t = 200$ s for S-E wind



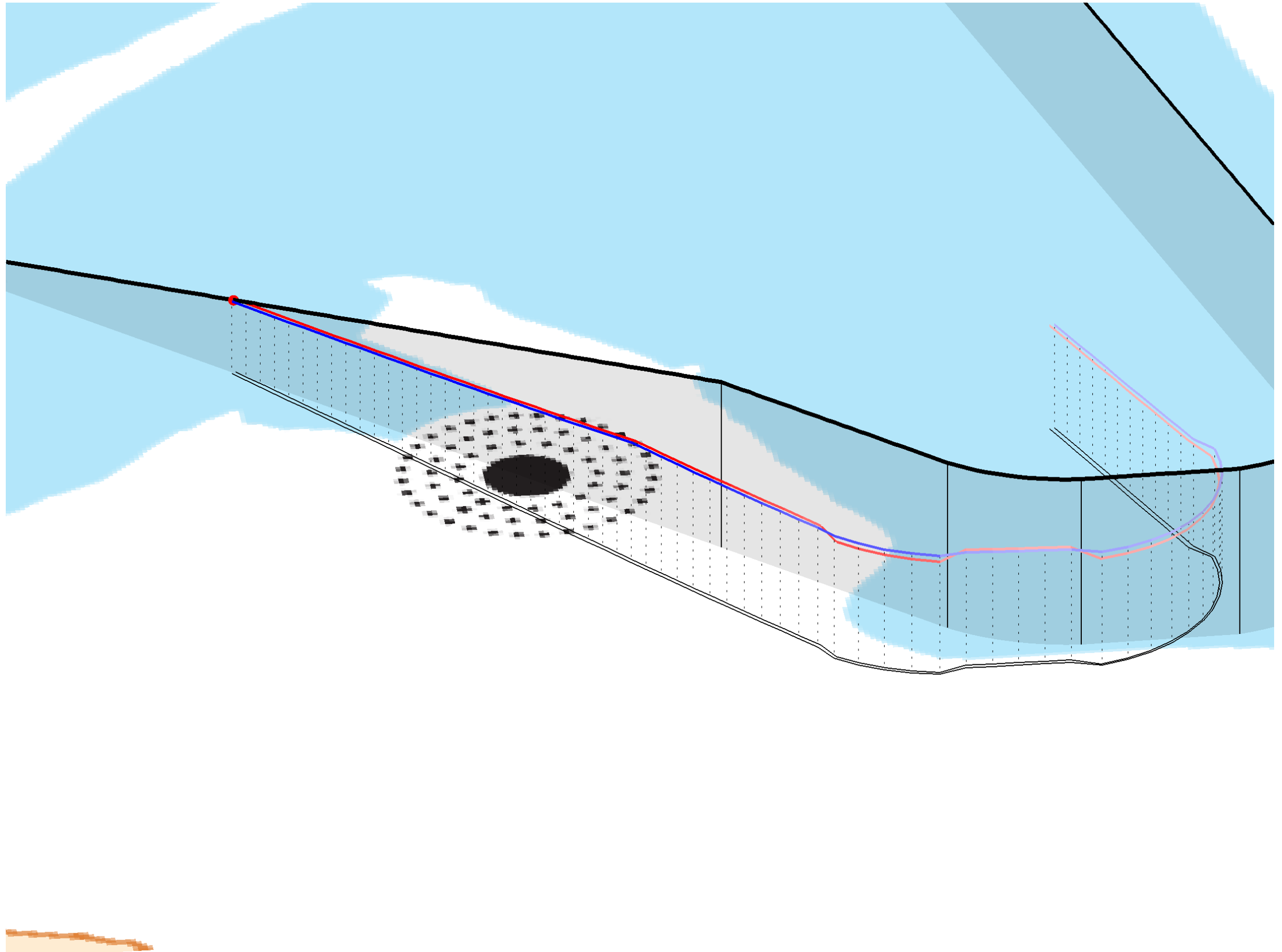
Wake at time $t = 200$ s for S-W wind



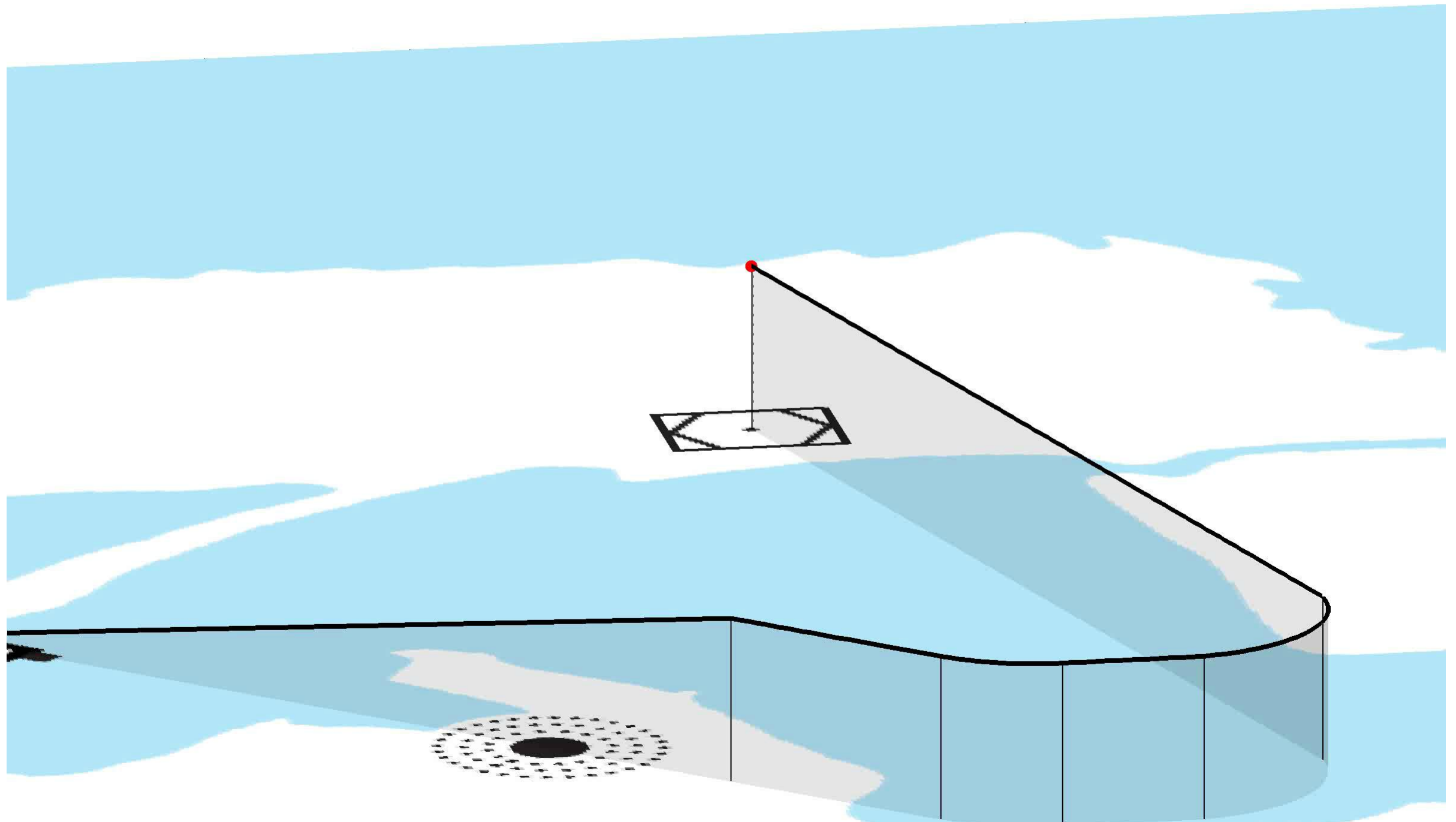
Wake at time $t = 300$ s for S-E wind



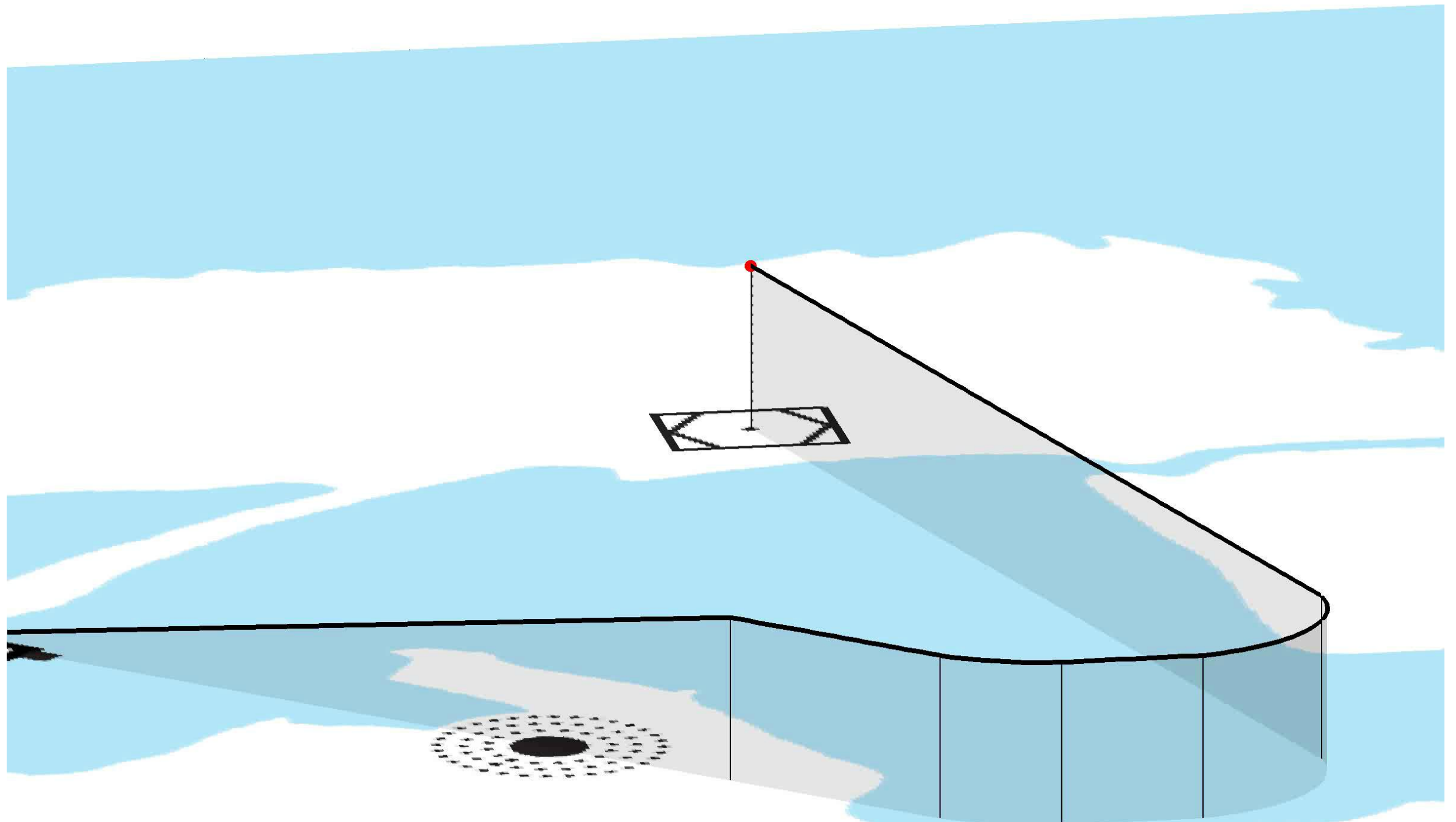
Wake at time $t = 300$ s for S-W wind



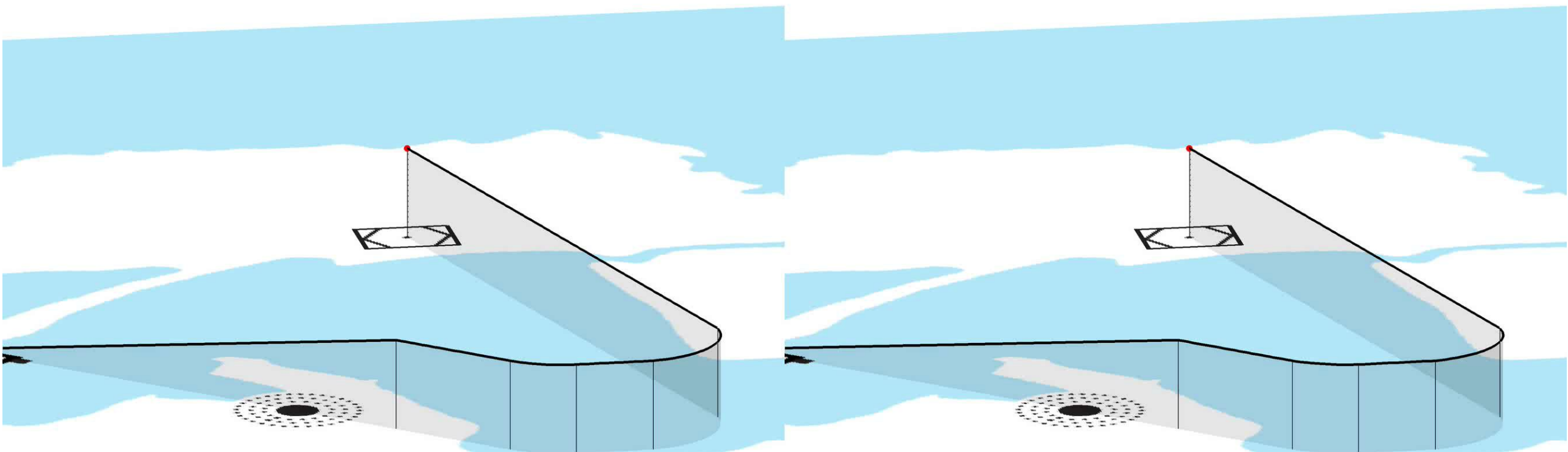
Animation for the S-E wind



Animation for the S-W wind

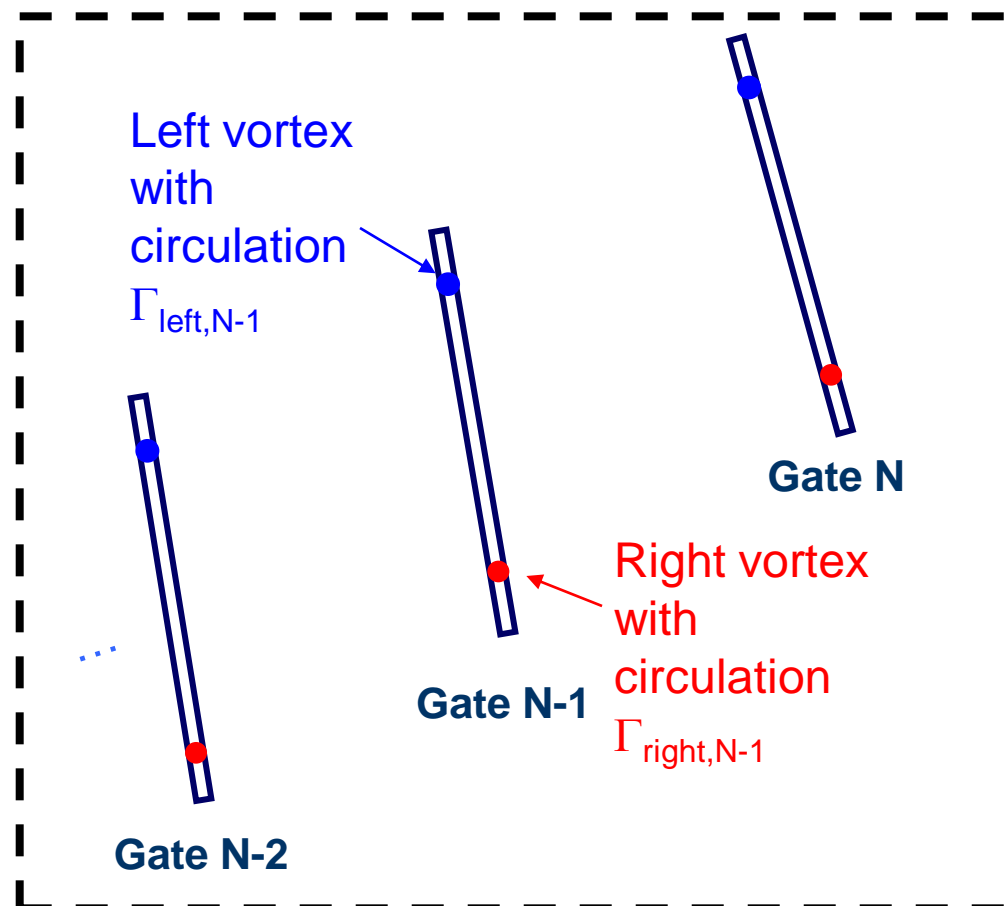


Animations for both cases

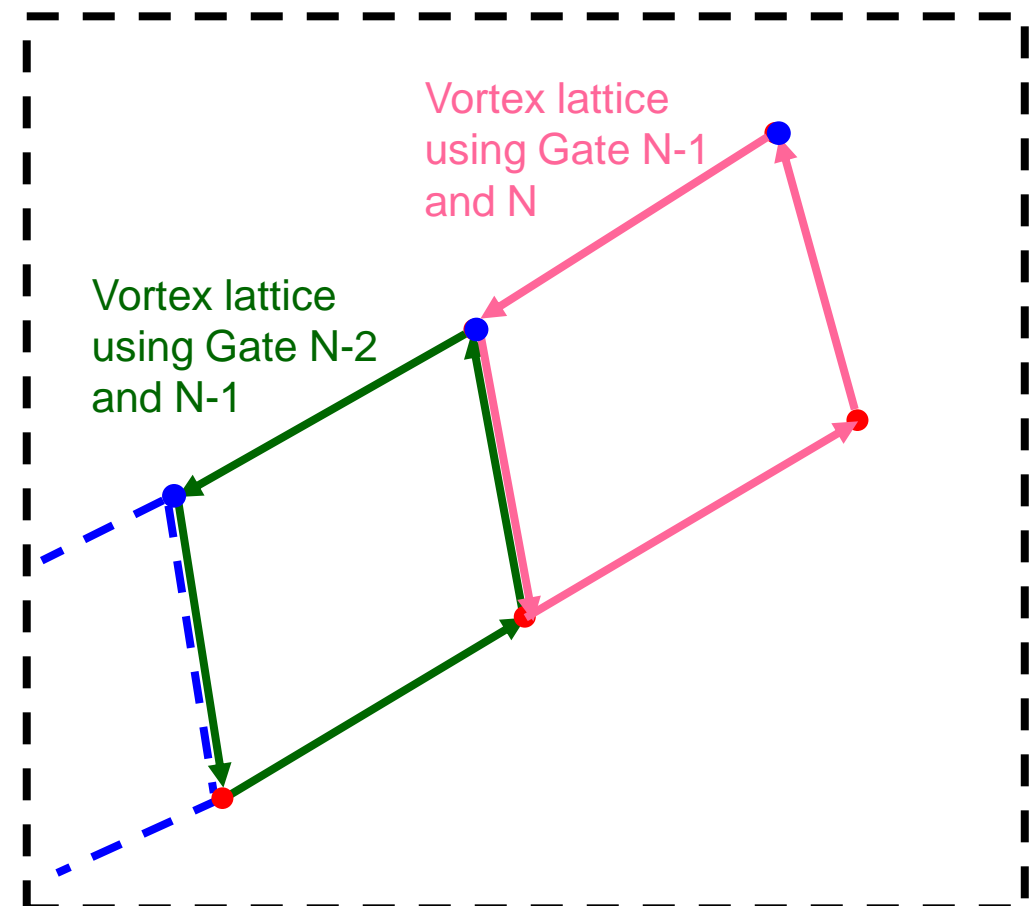


Induced velocity field evaluation using the WAKE4D-DVM results

- Between each pair of computational gates, an **equivalent vortex lattice** (i.e., vortex closed loop) is calculated with an average circulation.
- The total velocity field is equal to
 - the sum of the induced velocity of all “active” vortex lattices
 - plus the interpolated velocity induced by the secondary vorticity (if IGE).



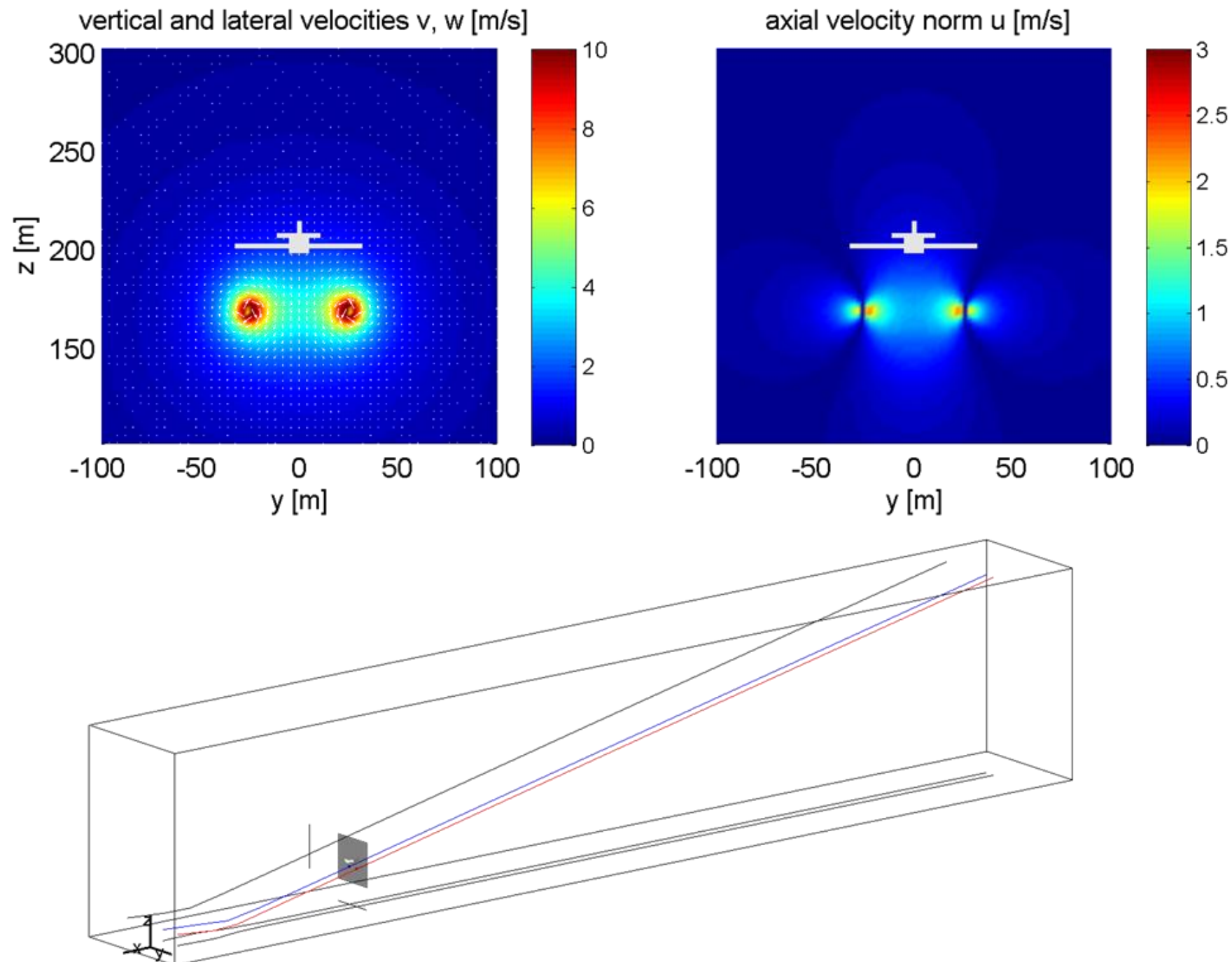
WAKE4D outputs at time t



Vortex lattice representation for velocity field evaluation at time t

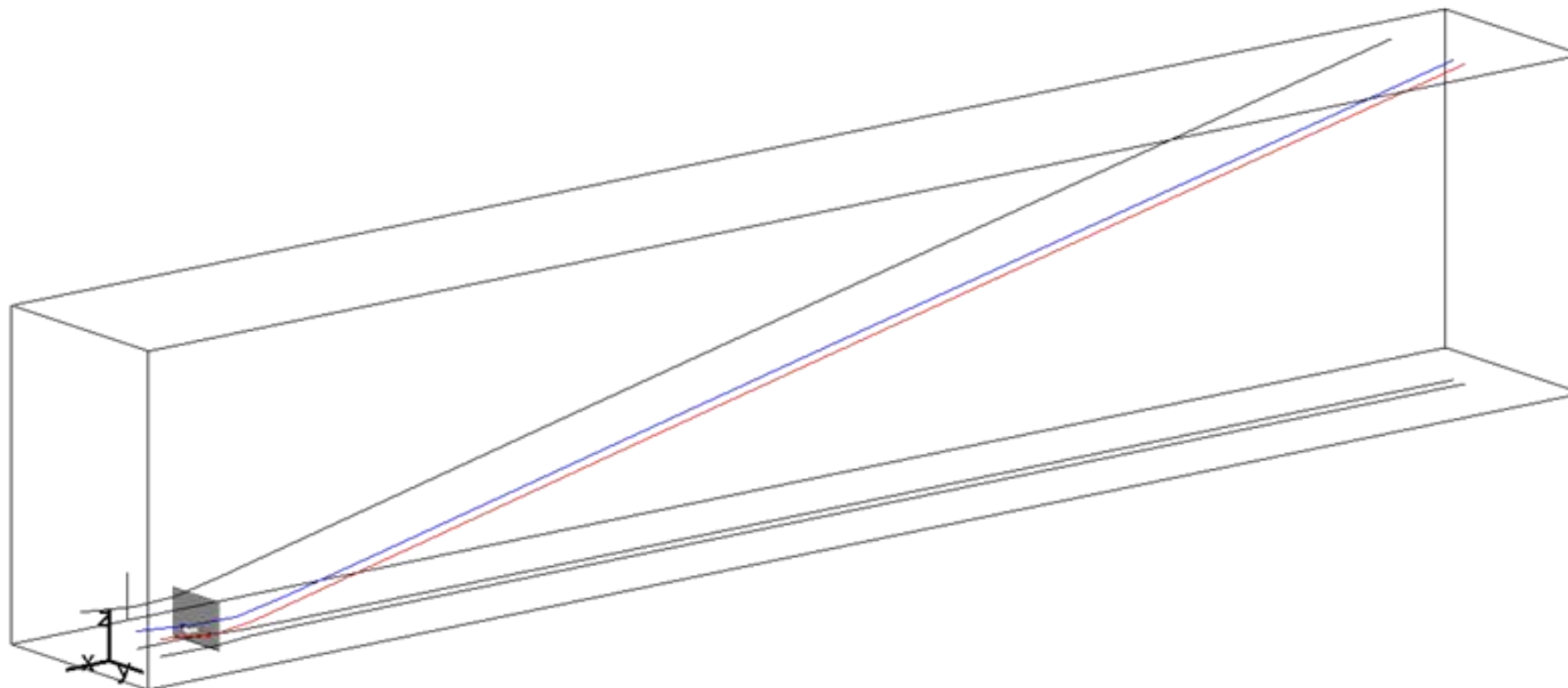
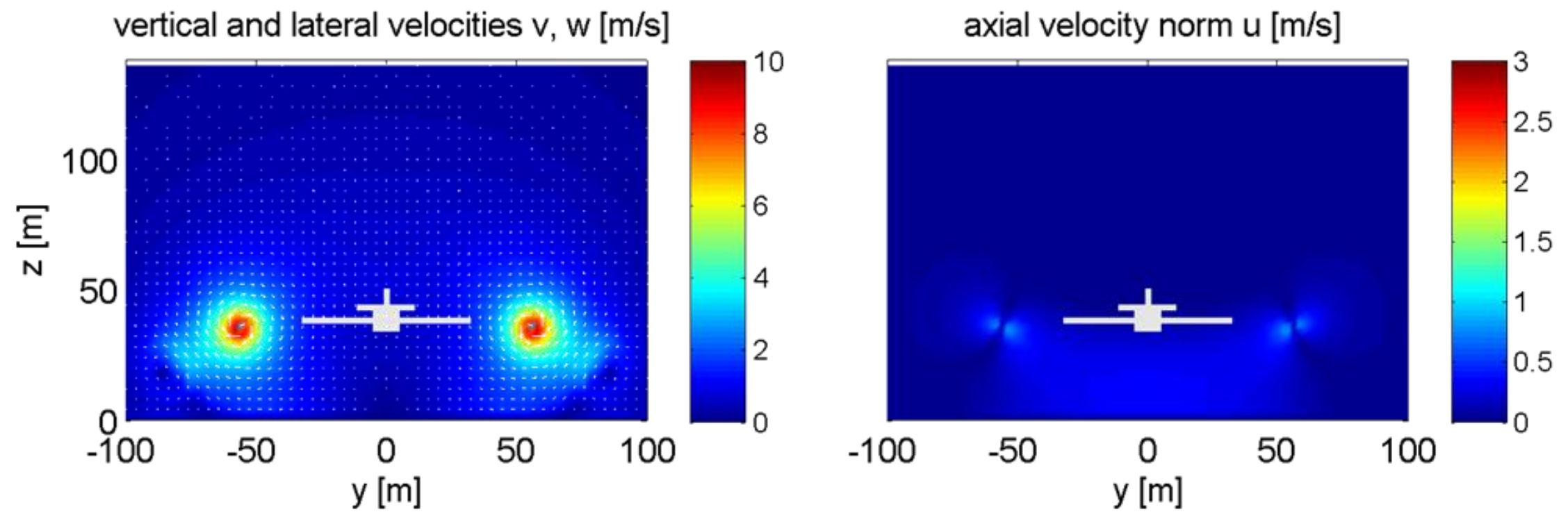
Induced velocity field evaluation using the WAKE4D results

- Evaluation of the velocity for **complex aircraft trajectory scenarios** (e.g., take-off, landing, turns, ...)
- **Real-time evaluation** of the induced velocity **at a hundred of points**.

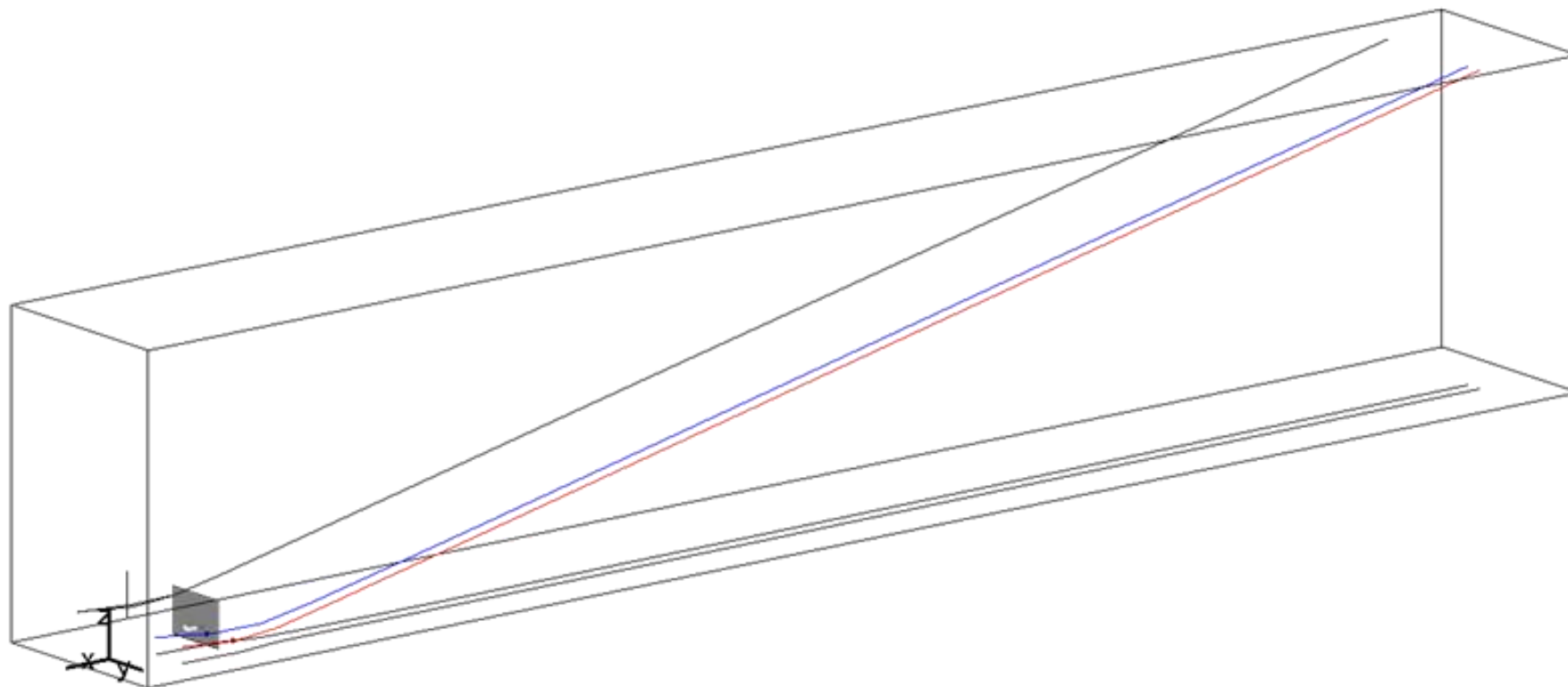
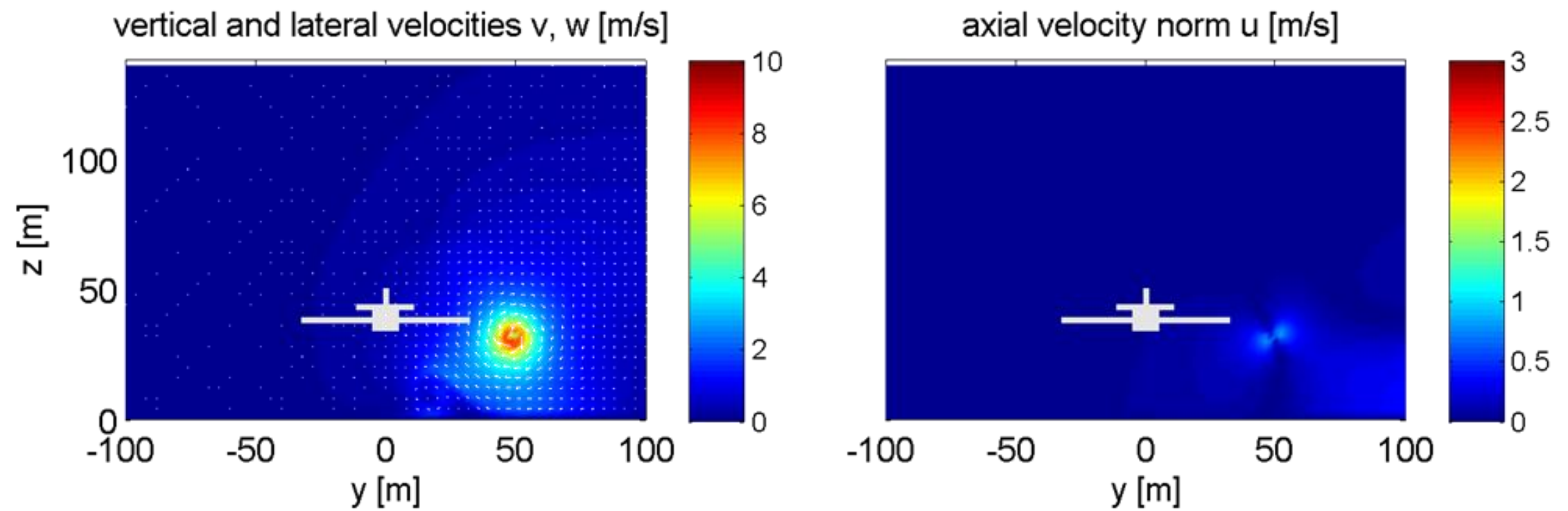


Example: Typical take-off of a large aircraft with headwind but no crosswind with a 90 s separation

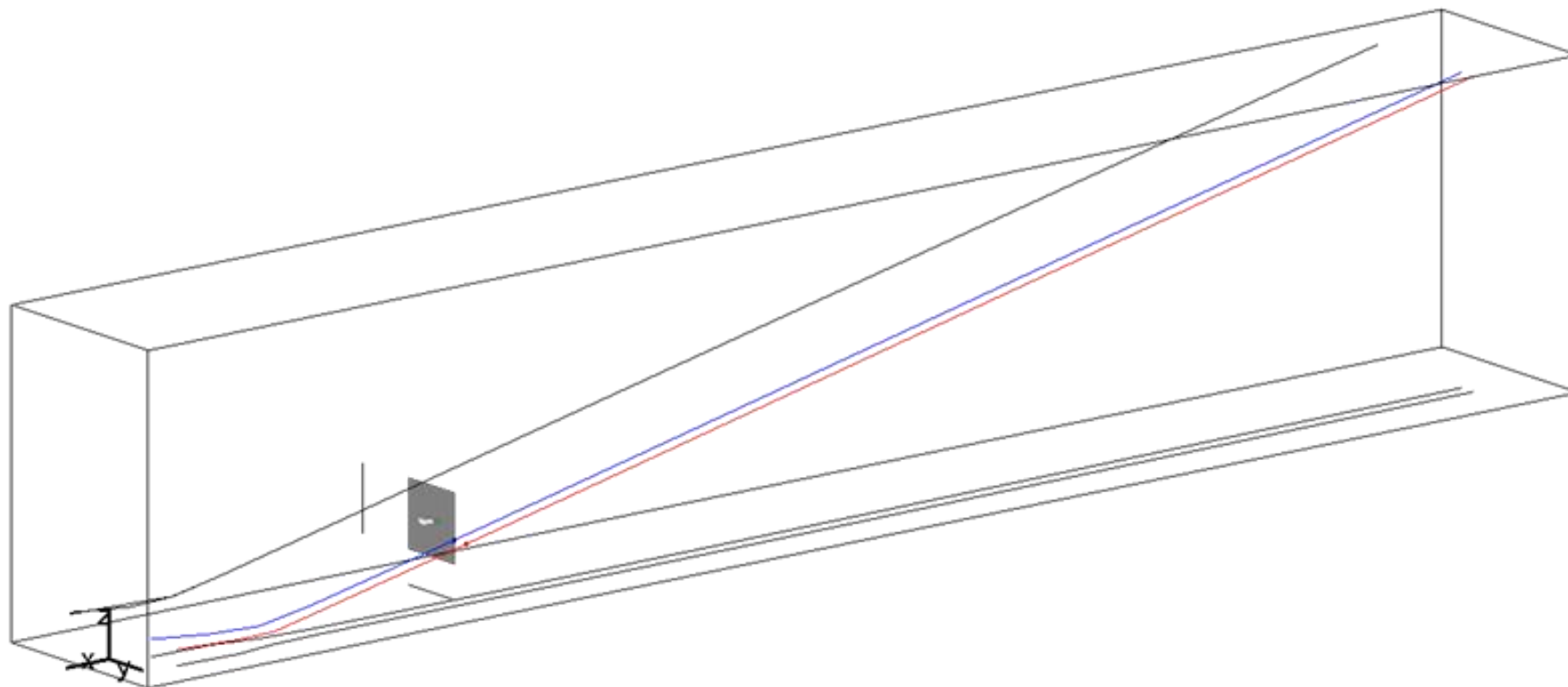
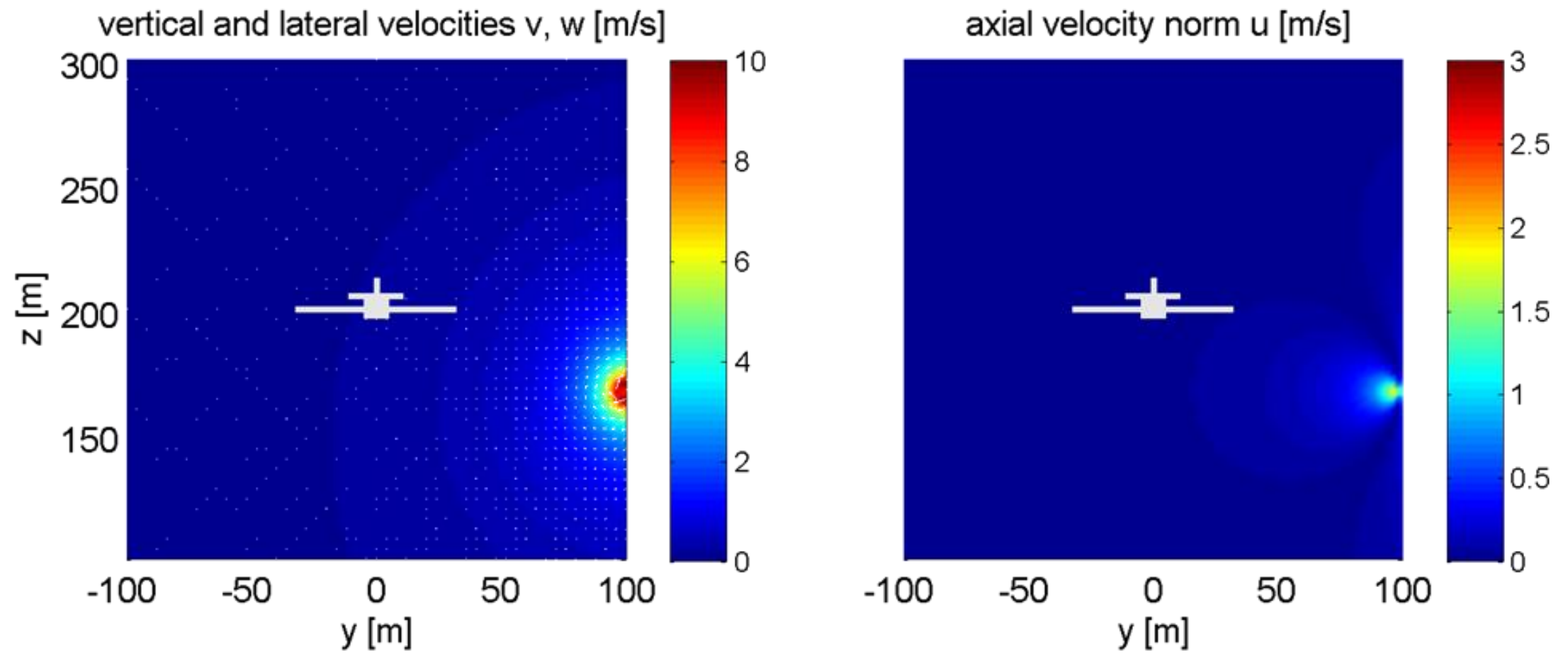
Typical take-off without crosswind with a 90 s separation



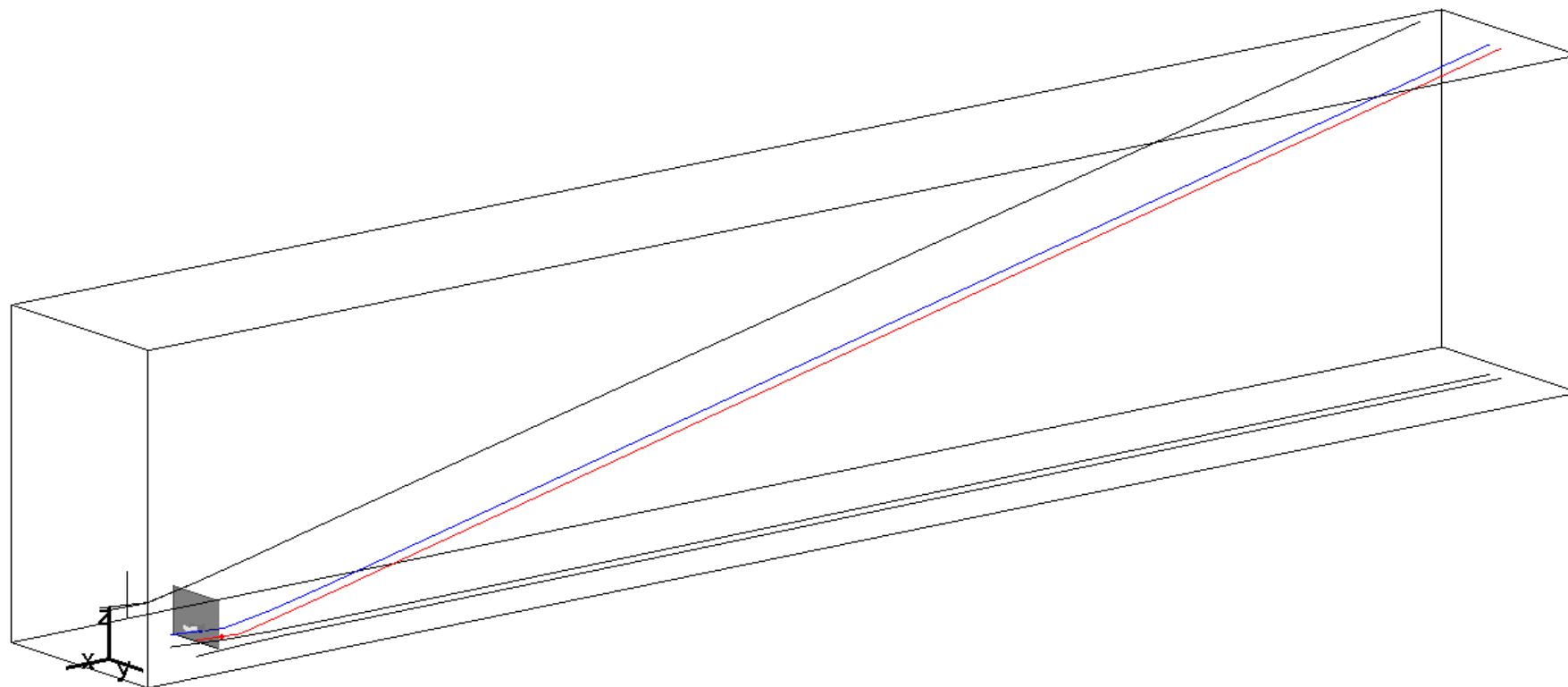
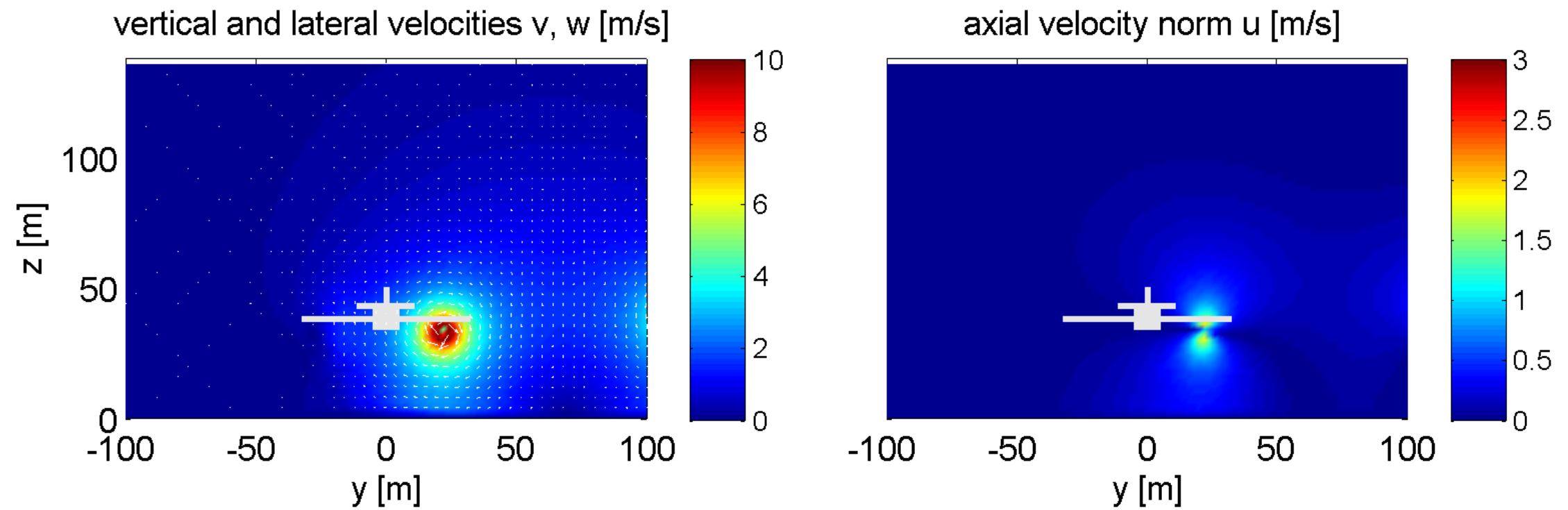
Typical take-off with low crosswind and with a 90 s separation



Typical take-off with low crosswind and with a 90 s separation



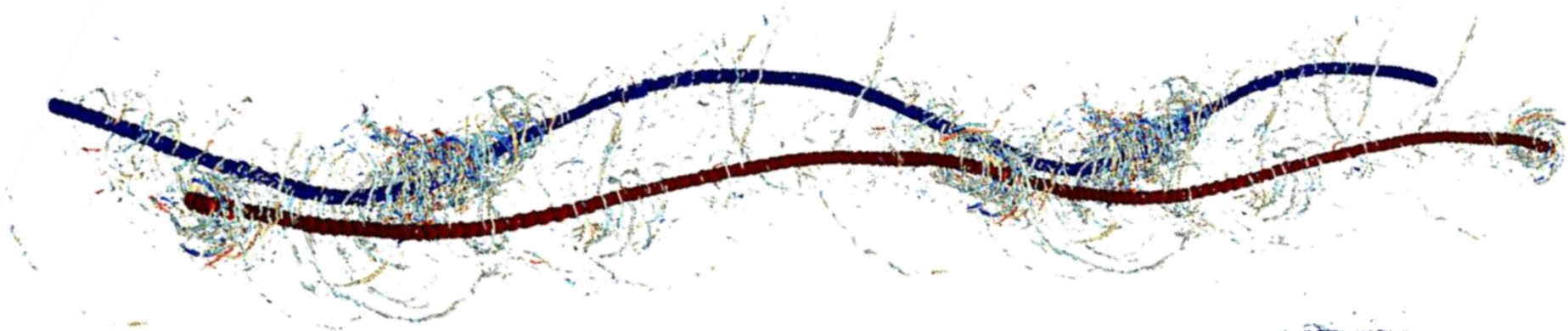
Typical take-off with low crosswind and with a 60 s separation



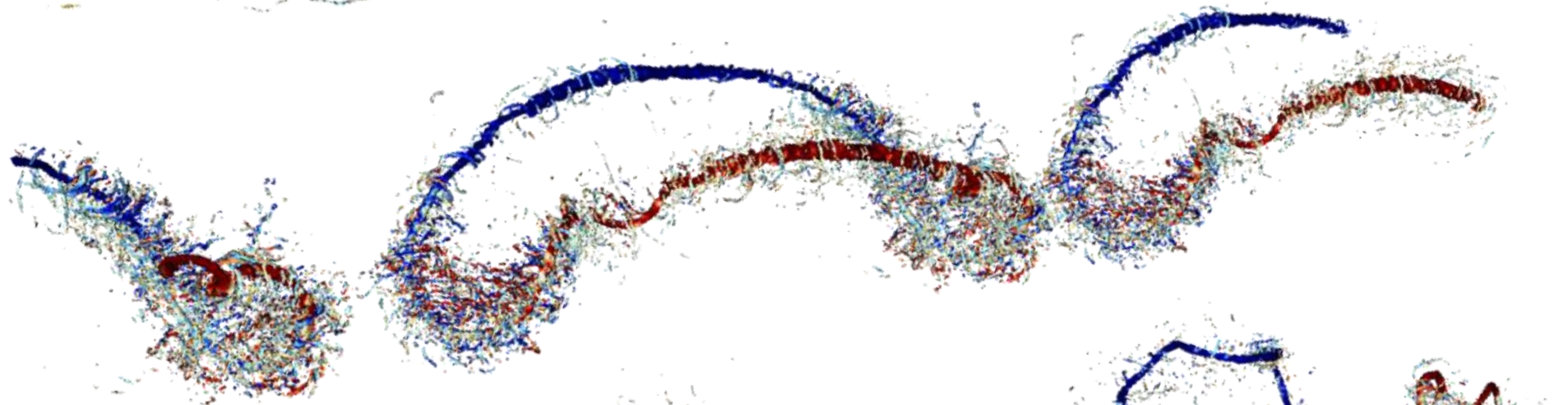
Crow instability



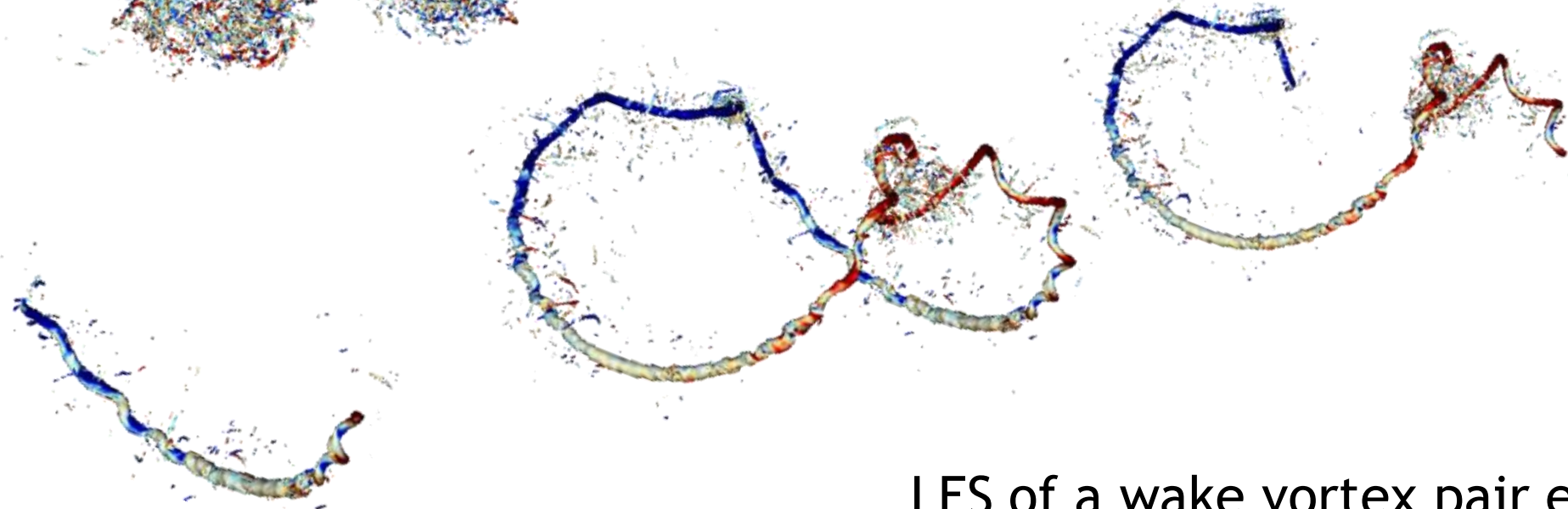
$\tau = 4.0$



$\tau = 5.0$



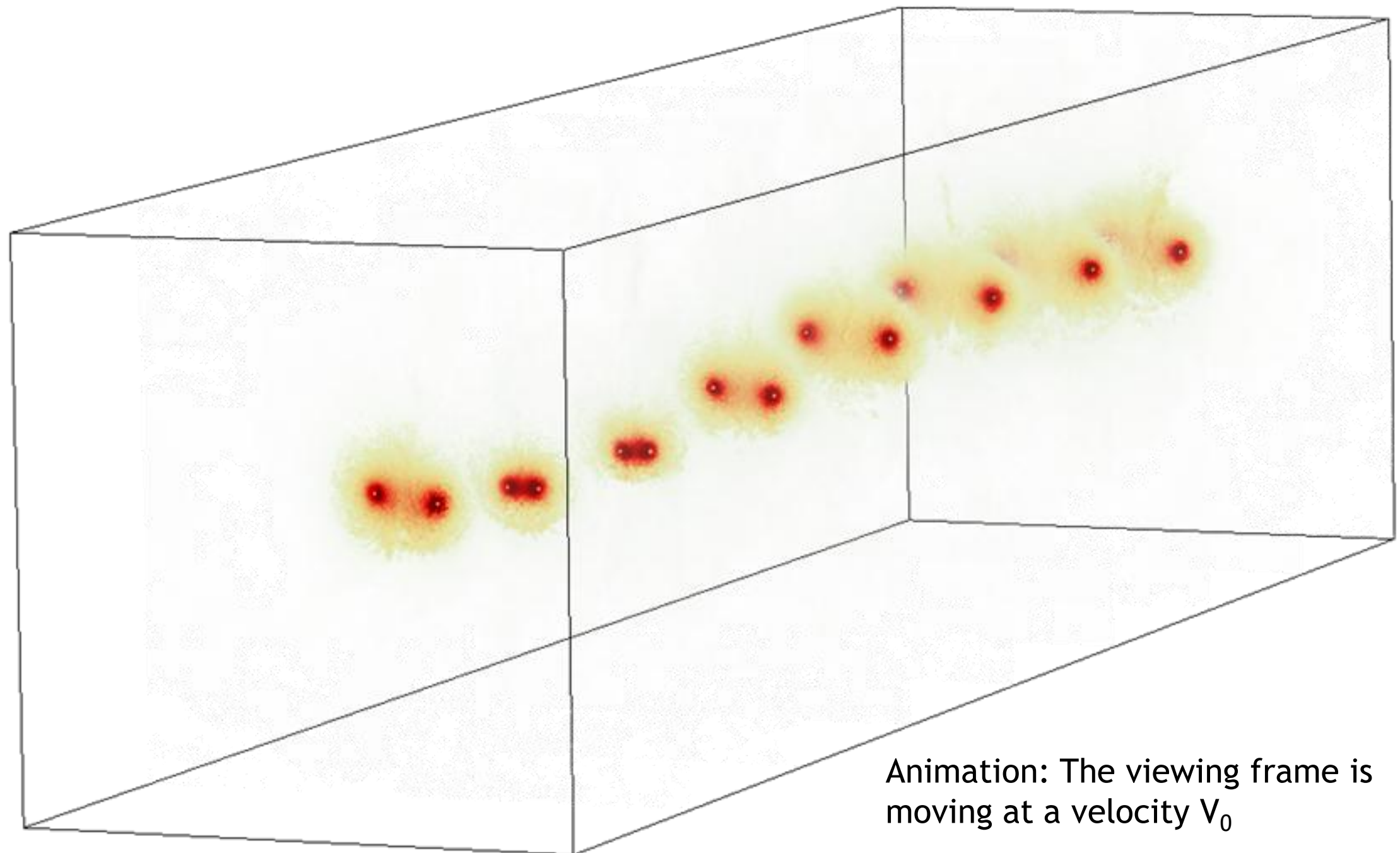
$\tau = 6.0$



LES of a wake vortex pair evolving in
a weakly turbulent atmosphere

Velocity norm in several planes (LES results)

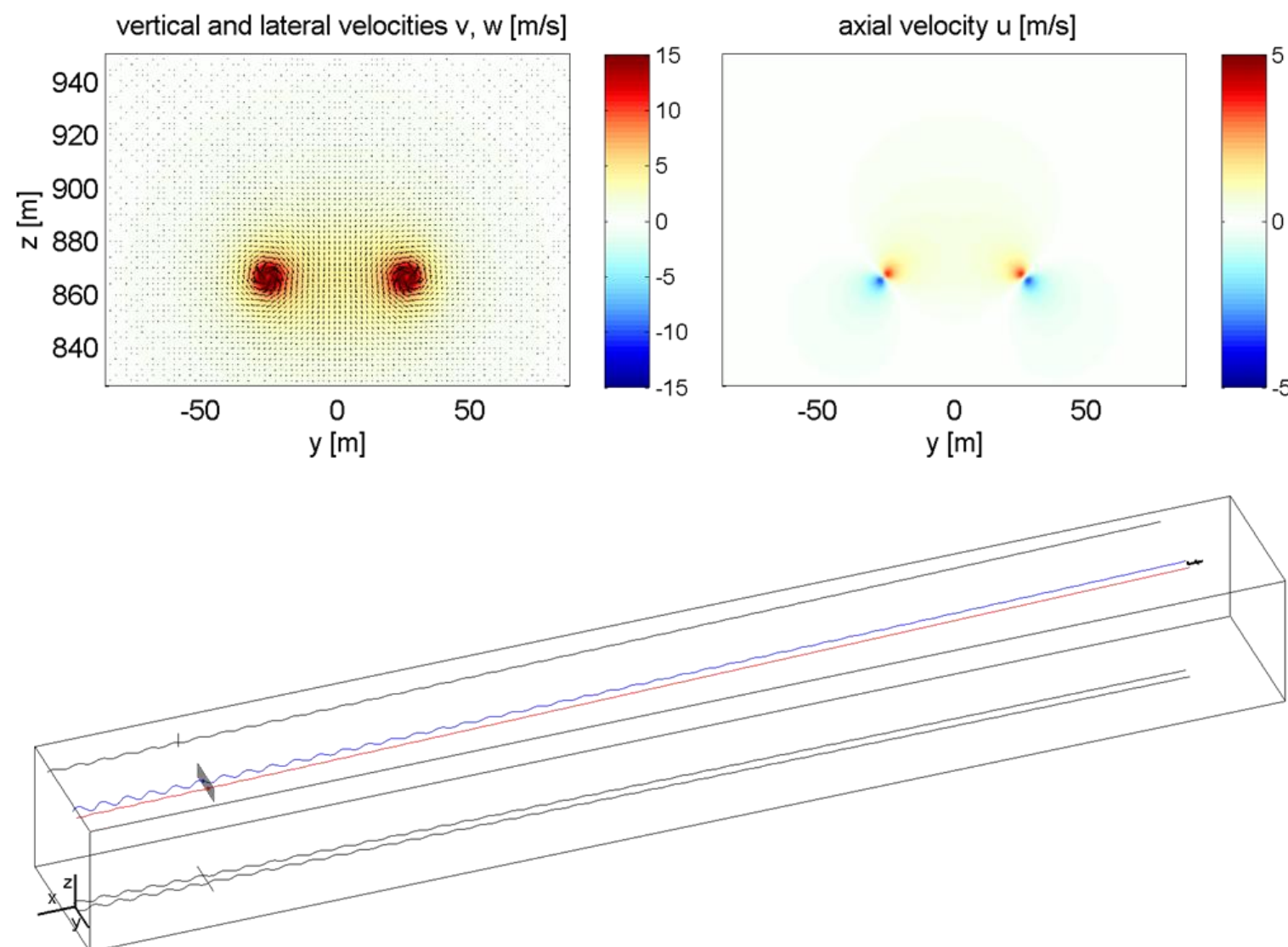
- Pseudo Spectral Code
- very high Re
- ~130 millions grid points
- ~ 1 month on 64 CPU's



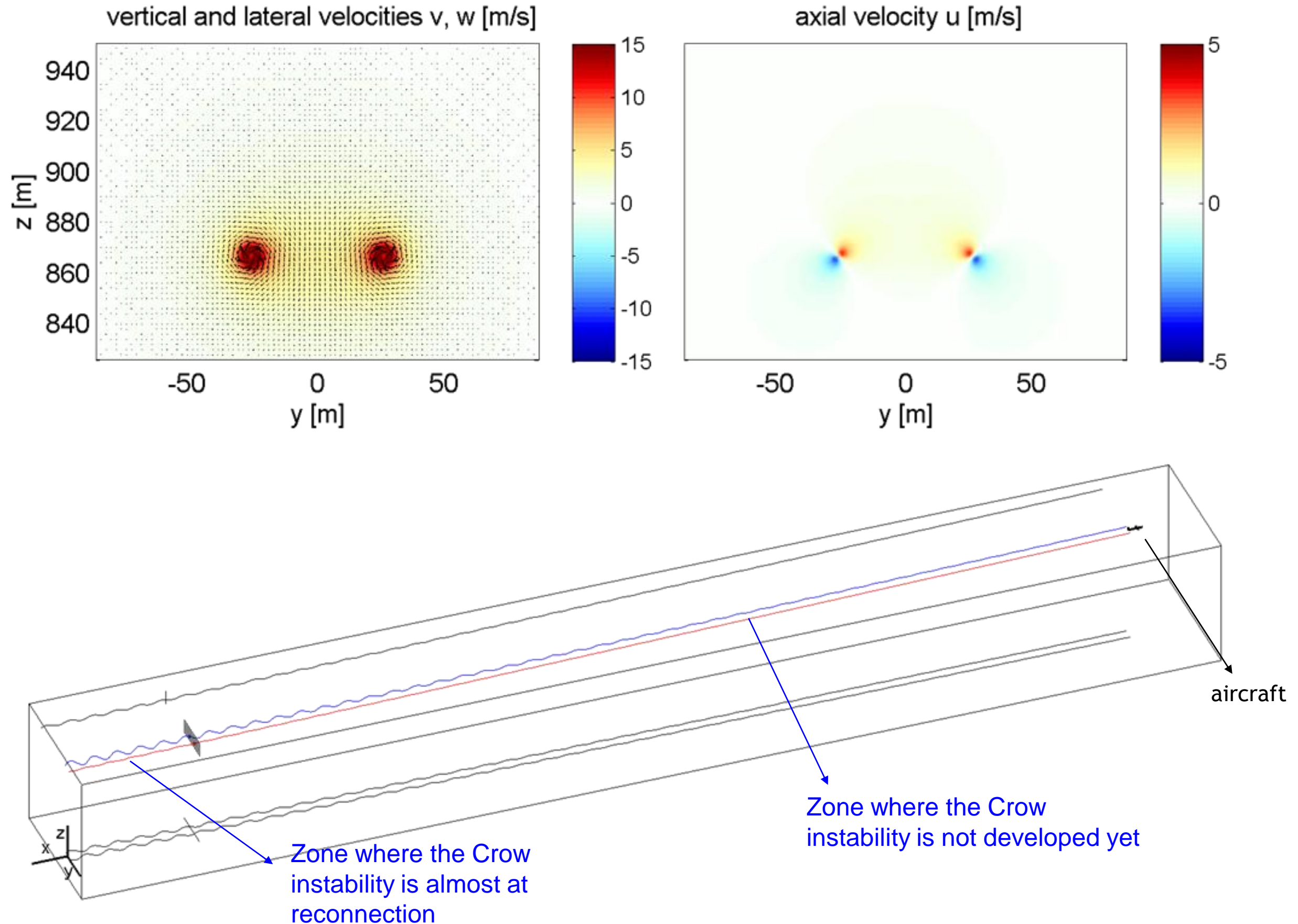
Animation: The viewing frame is moving at a velocity V_0

Induced velocity field evaluation using the WAKE4D-DVM results with Crow instability effect

- The deformed 3-D wake is rebuilt using the Crow instability amplitude as predicted in each computational gates.
- As post-processing, the **velocity field induced by the deformed vortices** can be evaluated, **in real-time**, using a vortex filaments approach with efficient Biot-Savart velocity evaluation
- This approach is useful for OGE conditions (also cruise) where Crow effects are important



Induced velocity field evaluation using the WAKE4D-DVM results with Crow instability effect : example



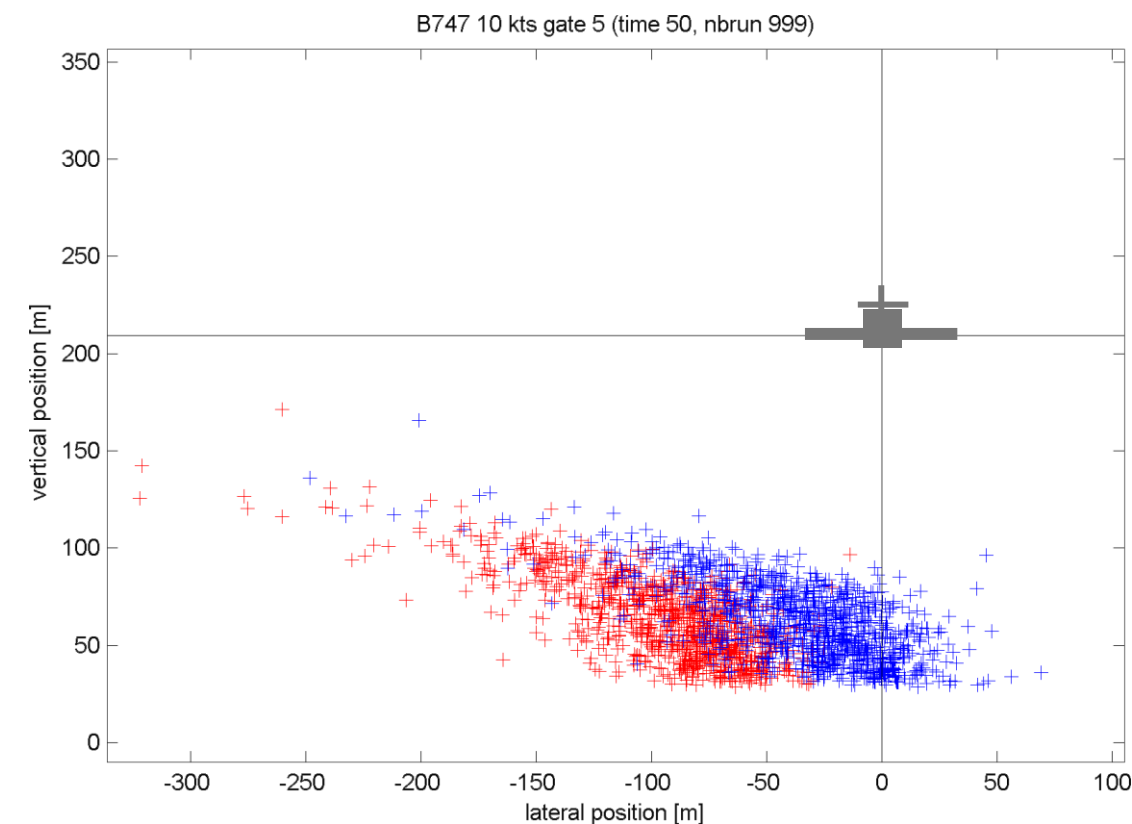
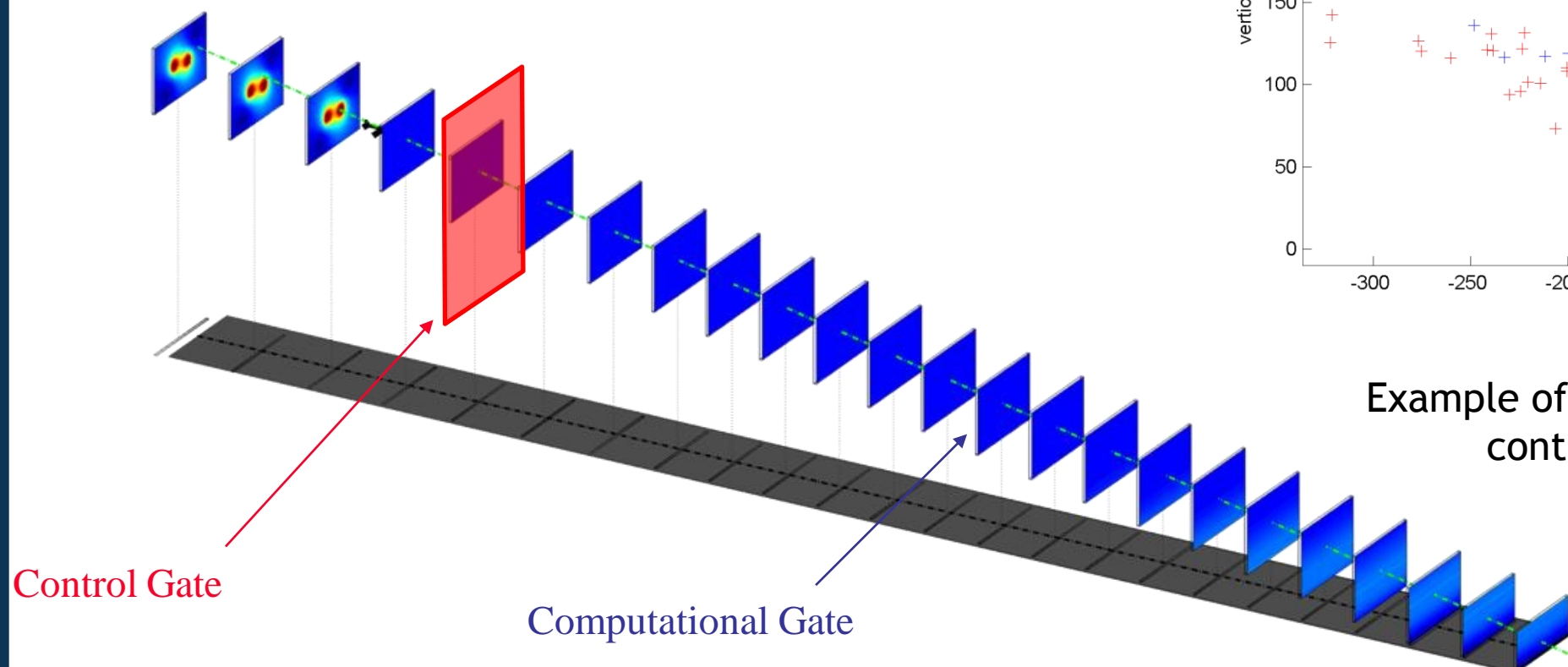
Investigation of a Time-based Separation (TBS) concept using the WAKE4D-PVM (Eurocontrol TBS project)

- New concept of operation applied to the arrival phase of flights with significant headwind.
- The current **distance-based separations (DBS)** are to be **replaced by time intervals**.
- The WAKE4D-PVM was used for a **comparative risk analysis** between DBS in still air conditions and TBS with headwind:
 1. Simulation of the 3-D wakes generated :
 - by **various aircraft** (with variations on their trajectories)
 - in **various wind** conditions (with variations)
 2. Interpolation of the results in several control gates along the flight trajectory.

Investigation of a Time-based Separation (TBS) concept using the WAKE4D-PVM (Eurocontrol TBS project)

In each of the control gates, computation of:

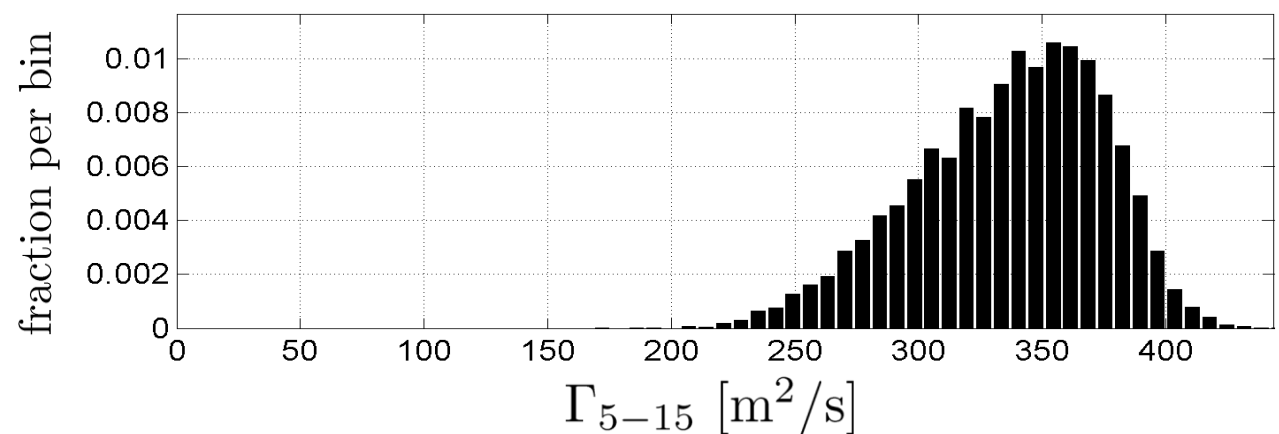
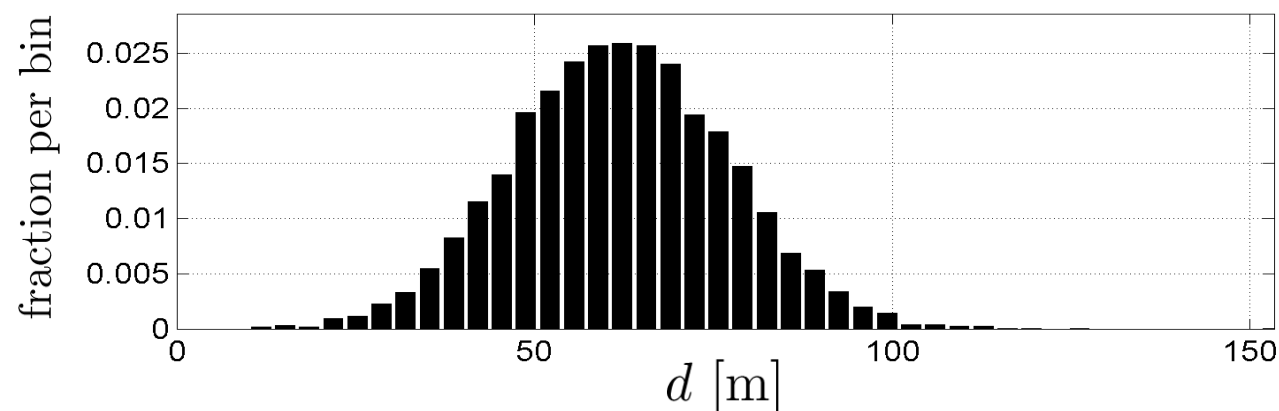
- the **distance between the WV** (both left and right) **and the following aircraft** (with uncertainties on its trajectory)
- the **PDF of the circulation** of the potentially encountered WV



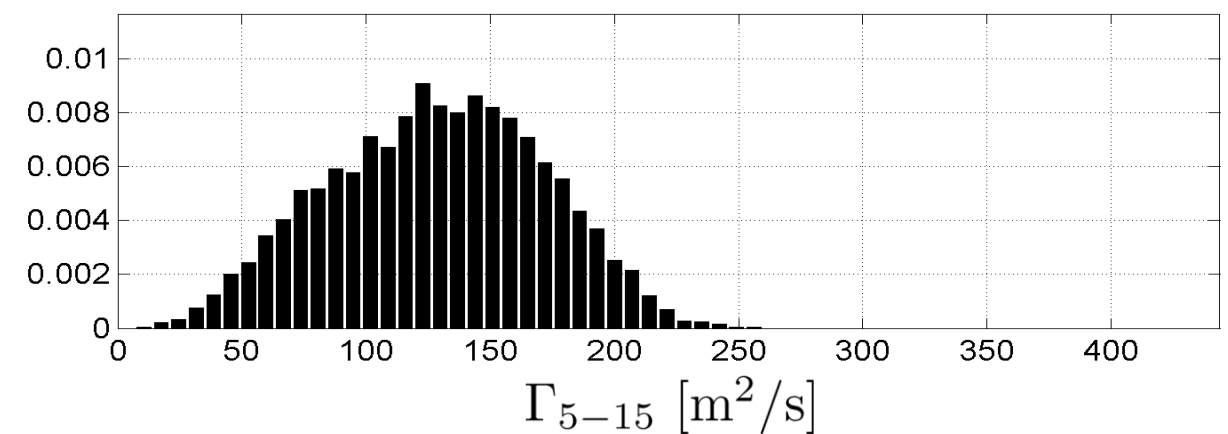
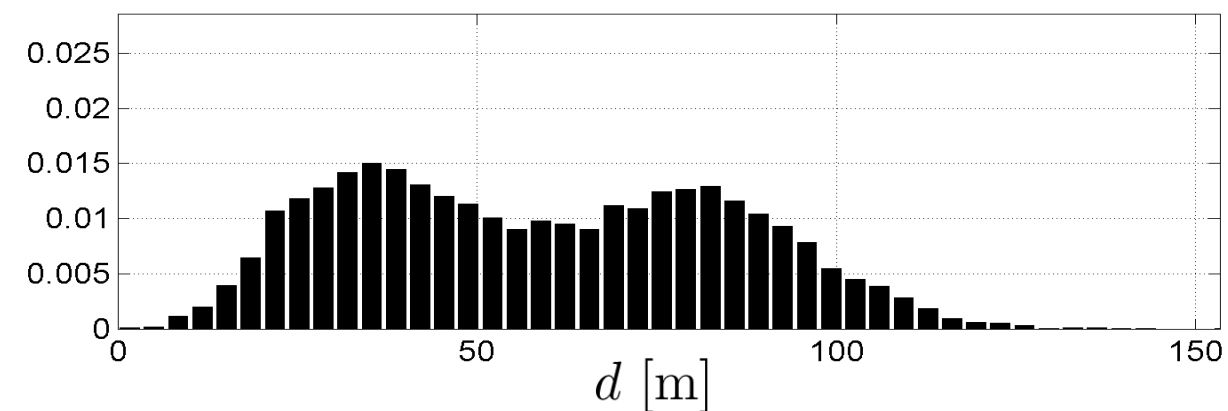
Example of predicted positions in a certain control gate at a certain time

Example of WAKE4D-PVM results for a 90 s TBS in a gate IGE (ILS altitude = 65 m)

Without wind



10kts headwind at 10 m height



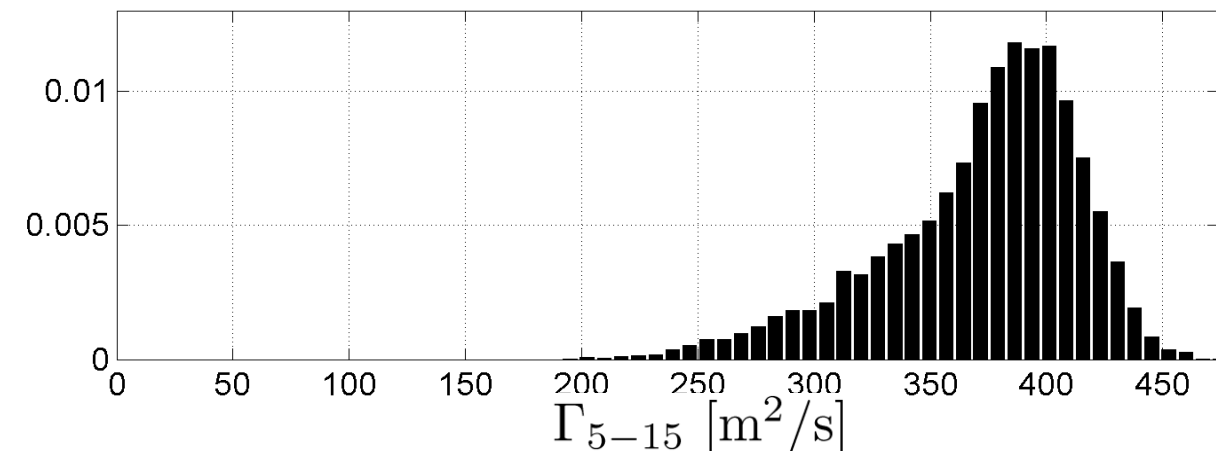
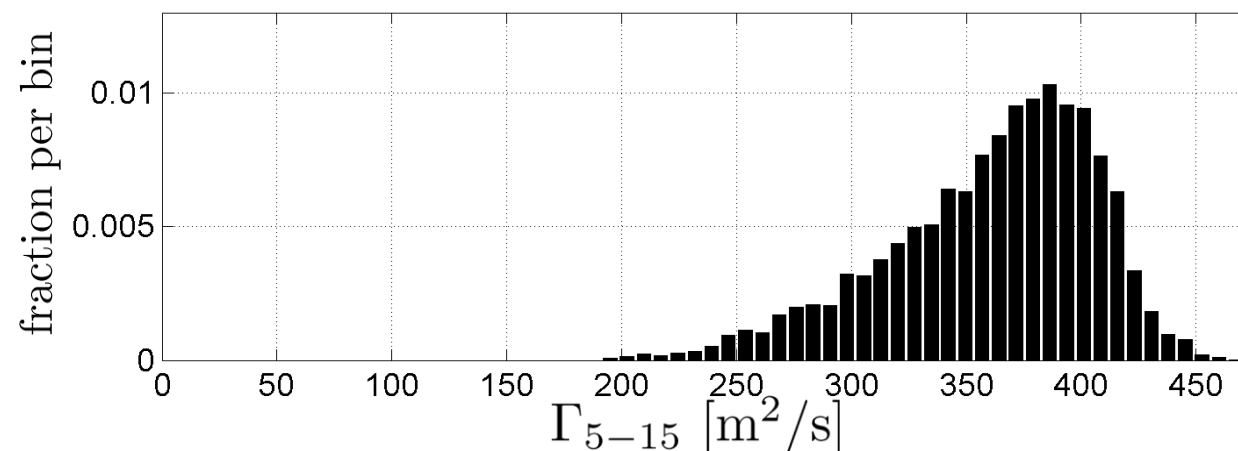
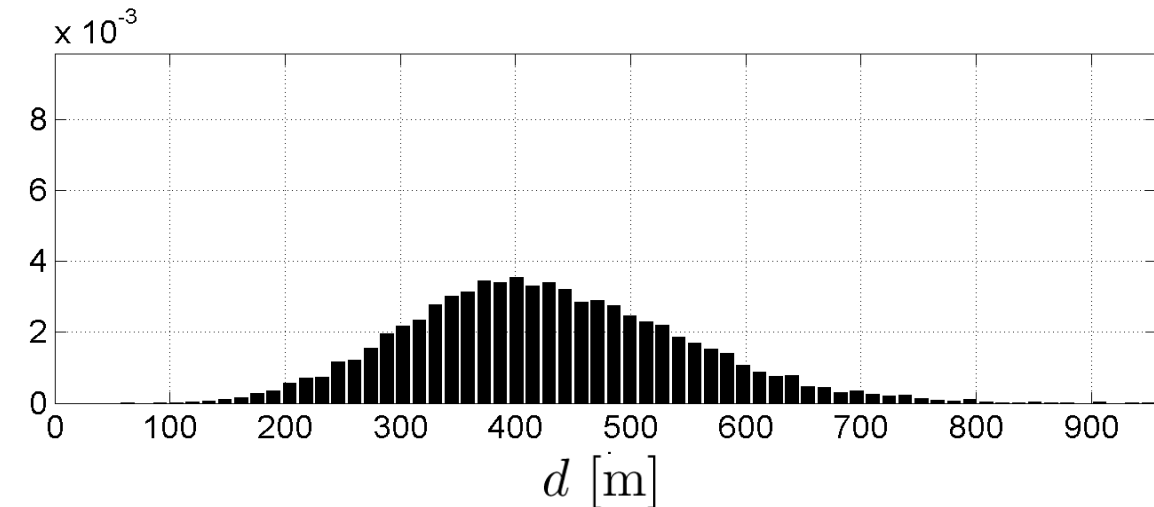
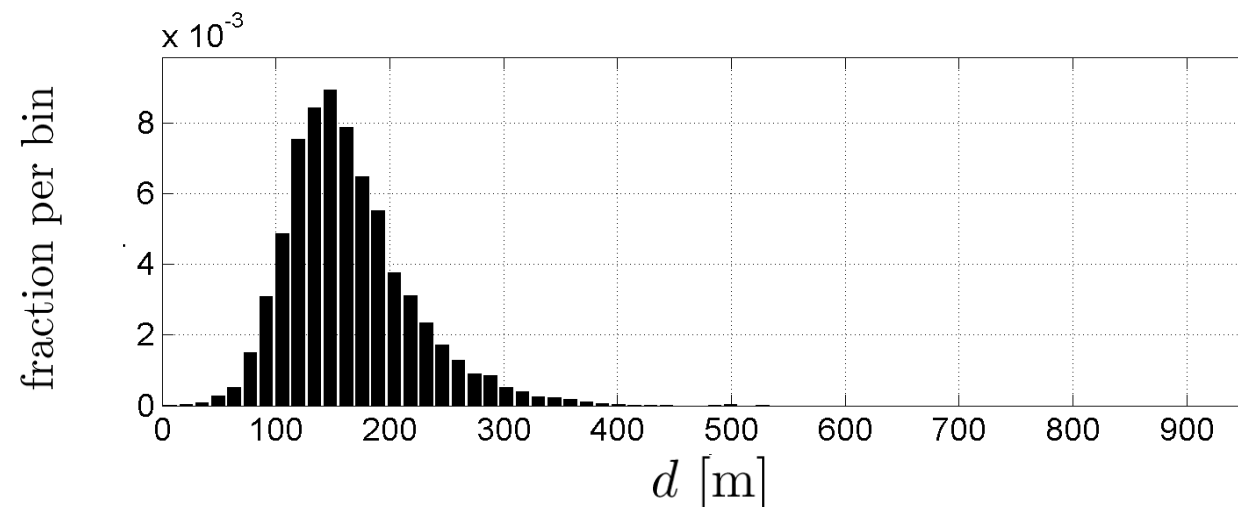
Close to the ground:

- the wind enhances the decay IGE
- the vortices rebound due to IGE and thus do not move away from the ILS

Example of WAKE4D-PVM results for a 90 s TBS in a gate OGE (ILS altitude = 915 m)

Without wind

10kts headwind at 10 m height

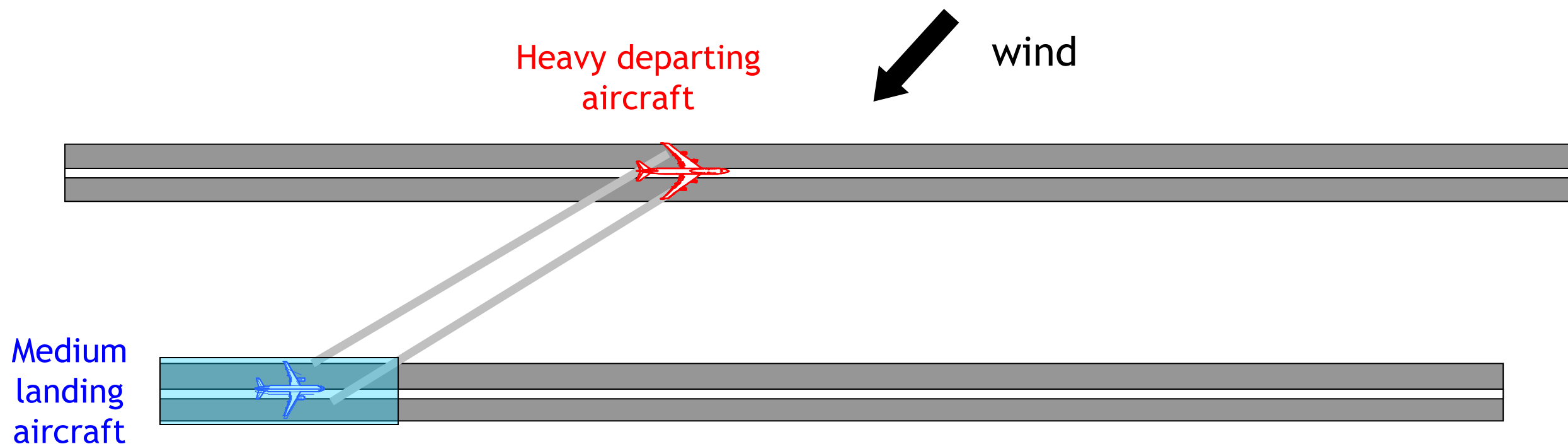


Away from the ground:

- the headwind tends to bring younger vortices in the control gate
- the crosswind component tends to move those WV away from the ILS

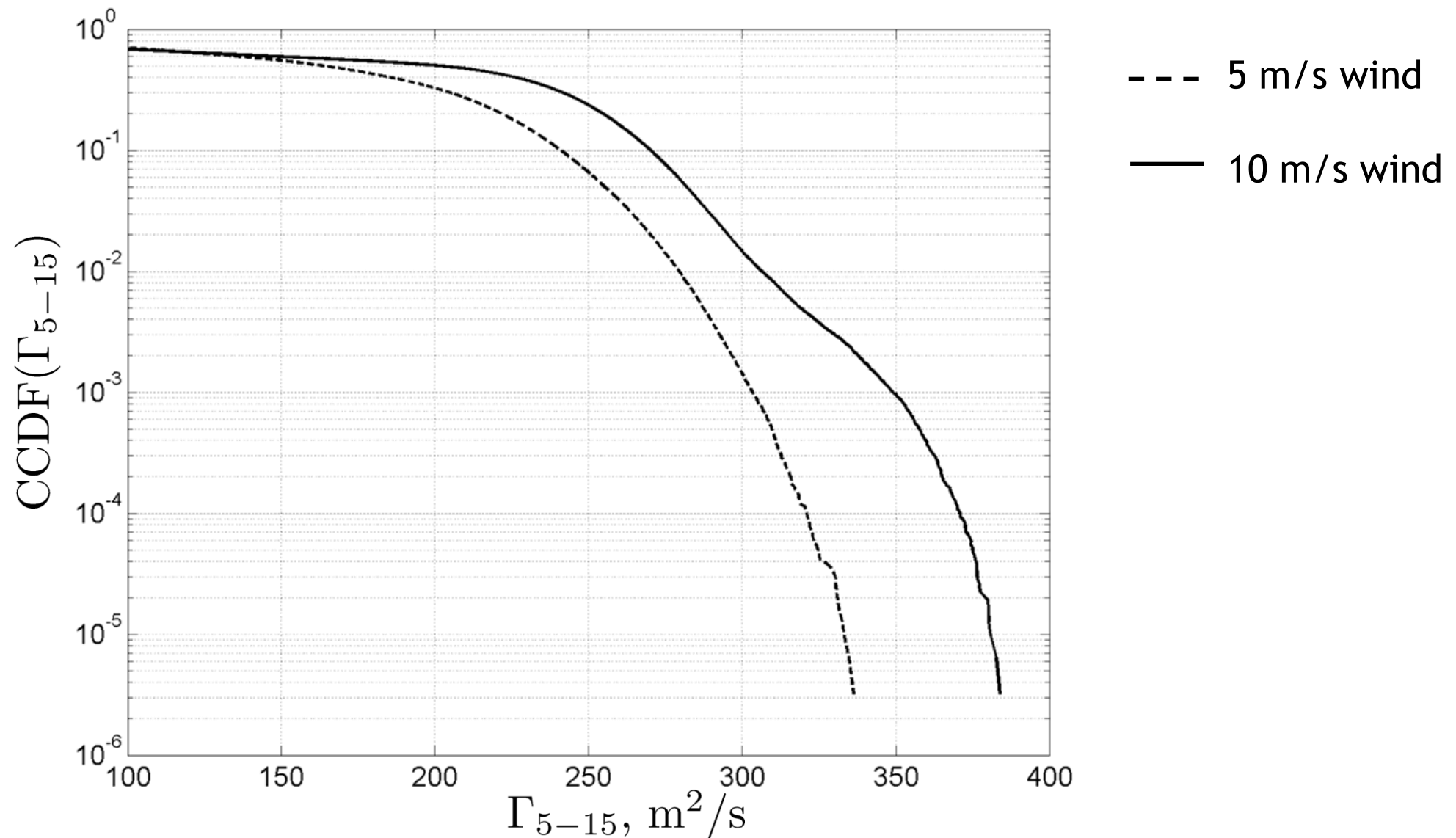
Investigation of the impact of the wind on potential WVE for a pair of parallel runways.

- **WV** generated by **heavy departing aircraft**
- WV **blown** by the wind in **the landing aircraft region** of the parallel runway (=“control box”).
- Two cases of wind are investigated:
 - a mean wind profile with a **5 m/s** wind at 10 m height
 - a mean wind profile with **10 m/s** wind at 10 m height.
- The **wind direction** is at **45 degrees** w.r.t. the runway.
- Variations of 0.5 m/s are assumed around these mean profiles, with a uniform distribution.



Comparison of the frequency-severity curves

The complementary cumulative density function ($CCDF(\Gamma_{5-15})$) is defined as the probability to observe a vortex with a circulation higher than Γ_{5-15}



- As expected, the **wind blows the vortices on the arrival runway**.
- A **higher wind** leads to younger potentially encountered vortices with **higher circulation**.
- Note : the wind direction was here chosen so that the vortices always come in the arrival zone;
=> the probability to observe a vortex with any circulation is almost unity.

Conclusions

- The WAKE4D and its sub-components, DVM/PVM:
 - constitute a **well mature** and **operational WV predictor**
 - enable to simulate realistically complex WV behaviors depending on met conditions and a/c trajectory
- Concerning the DVM
 - The **physical models** have been established, improved, calibrated, assessed and used for the past 15 years in several projects => can be considered as **reliable**
 - The software is **modular** and can easily be modified (e.g., to use other models)
- Concerning the PVM
 - The **real Monte-Carlo** approach enables to perform **relevant statistical analysis**, taking into account input variations and uncertainties
 - The **resampling technique** enables to obtain conservative statistics while limiting the number of runs
- Concerning the WAKE4D
 - It is a usable WV predictor for **off-line fast-time studies** (e.g., potential WVE, reconstruction of accidents, and risk assessment)
 - It enables to compute in real-time, as post-processing, the **induced 3-D velocity field**
 - It is usable as a WV predictor for practical **real-time operational system**